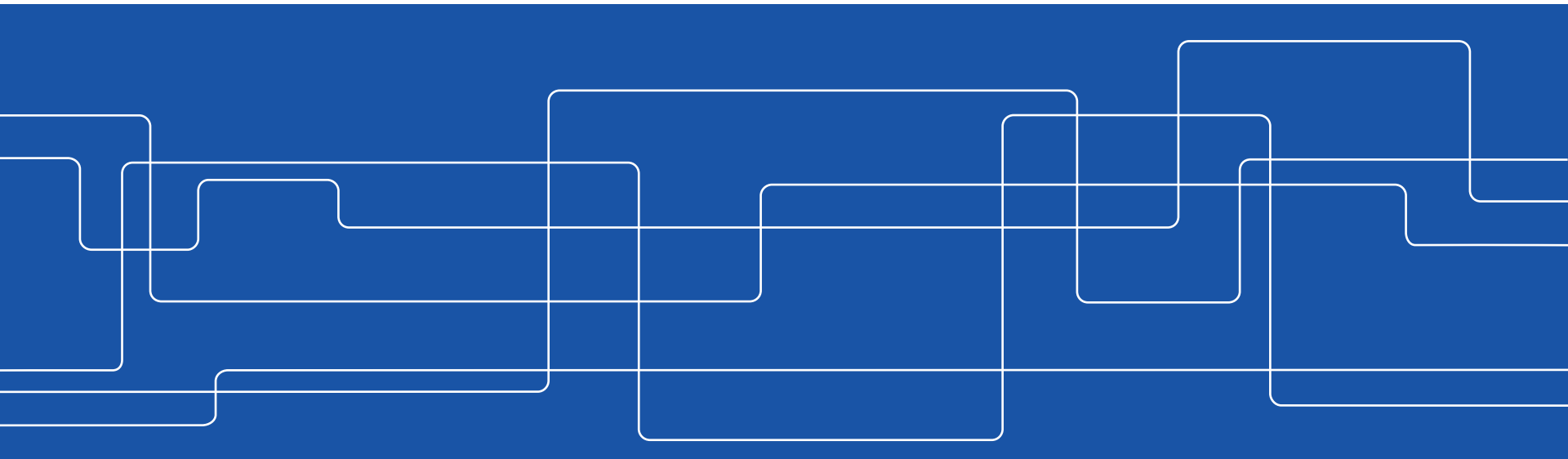


Steam Turbine Technology

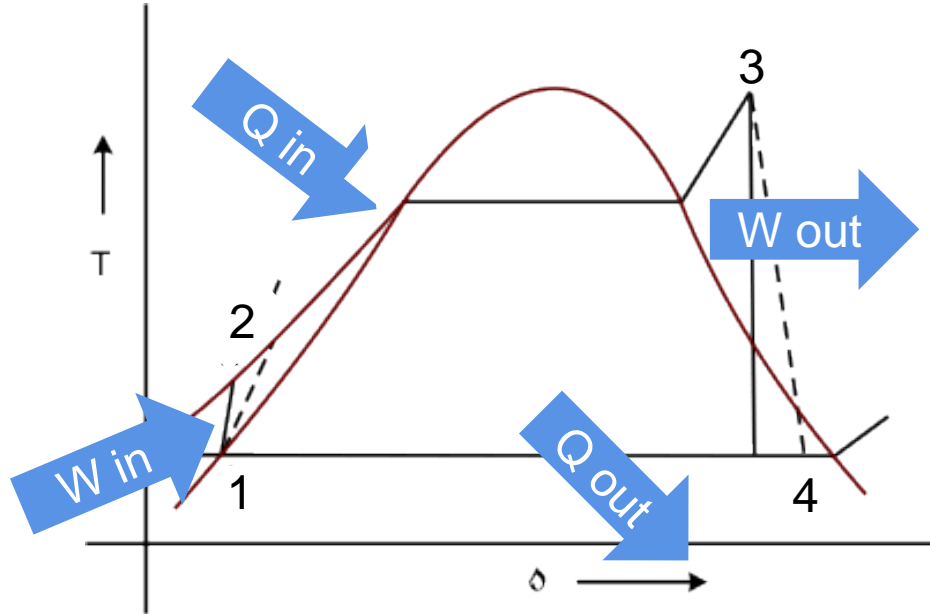
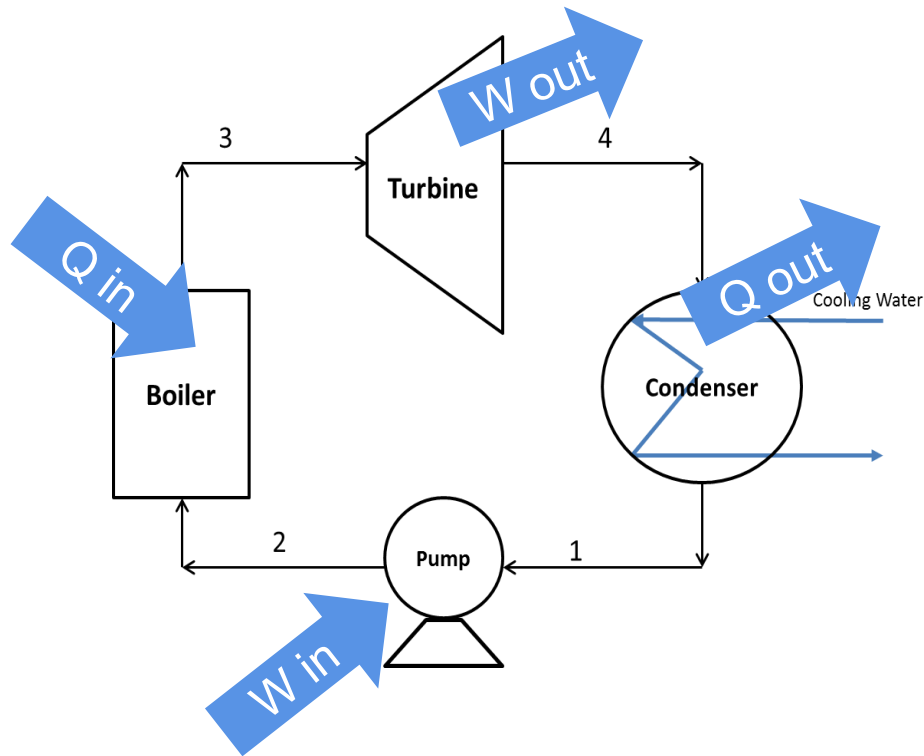
Design aspects of major components

MONIKA TOPEL, PHD

Based on: MARKUS JÖCKER, PHD, Siemens Industrial Turbomachinery



Rankine Cycle

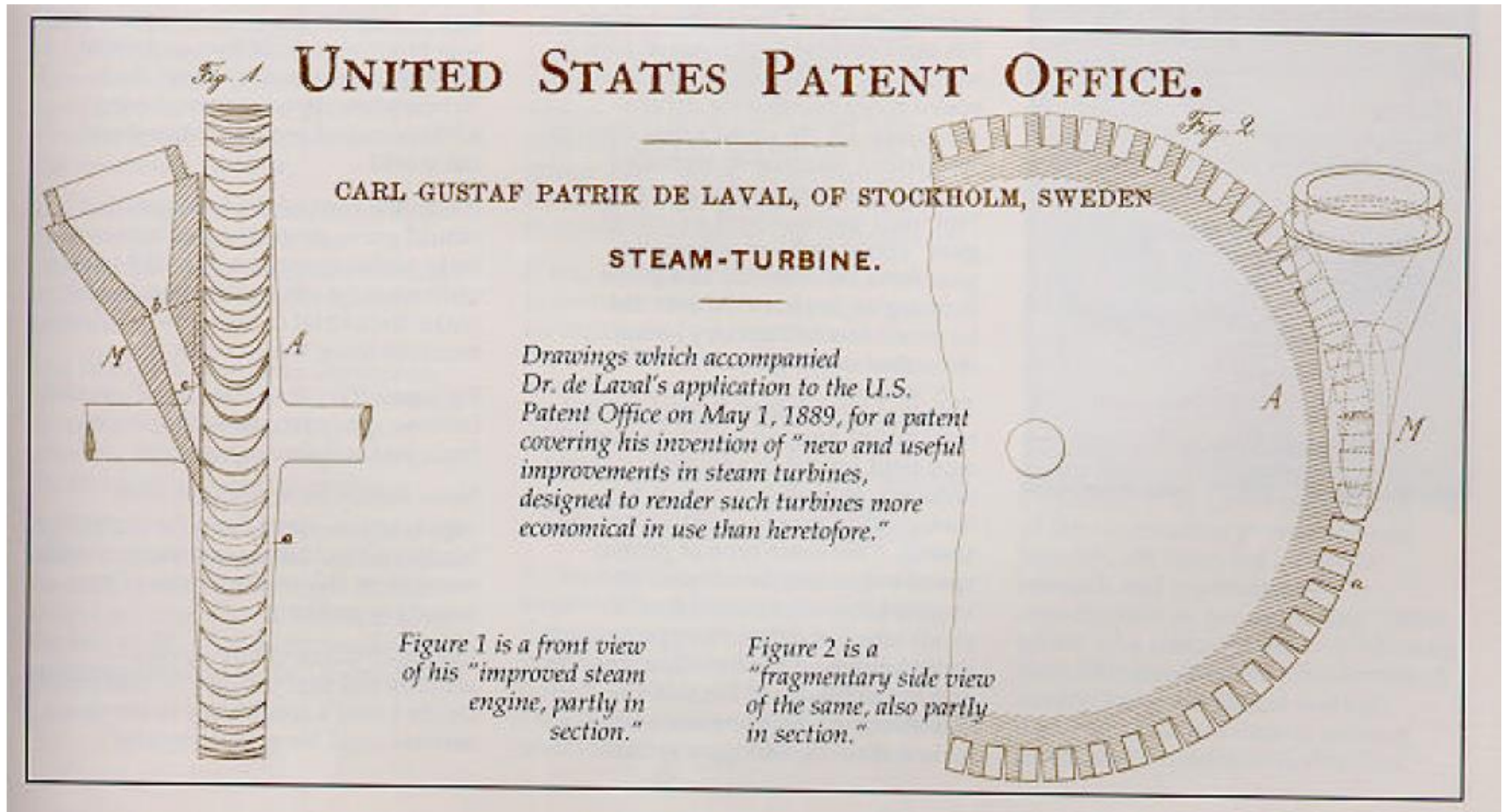


- 1-2 Isentropic compression in a pump
- 2-3 Isobaric heat addition in a boiler
- 3-4 Isentropic expansion in a **turbine**
- 4-1 Isobaric heat rejection

History – Significant milestones

- 1850** **Clausius-Rankine develops practical process**
- 1883** First reaction turbine by **deLaval**
- 1884** First multi-stage reaction turbine by **Parsons**
- 1888** First impulse turbine by **deLaval** and first power station to produce electricity with a turbine (**Parsons**)
- 1894** The counter rotating **Ljungstrom** turbine (STAL)
- 1903** Turbine theory and calculation methods by **Stodola** and the **Mollier** diagram
- 1920+** 30 MW units with 20bar/325 degC **eff<0.18**
- 1934** First four flow LPT on a single shaft **eff=0.275**
- 1937** Steam turbines with 120 bar/500degC **eff=0.36**
- 1964** First combined cycle
- 1965** 250 MW units with reheat **eff=0.45**
 First supercritical unit with double reheat **eff=0.48**
- 1970+** 500-900 MW coal fired units and 1400 MW nuclear
- 1988+** Combined cycle starts gaining popularity

De Laval's patent for impulse turbine from 1889



Today: Challenges

Dominant worldwide!

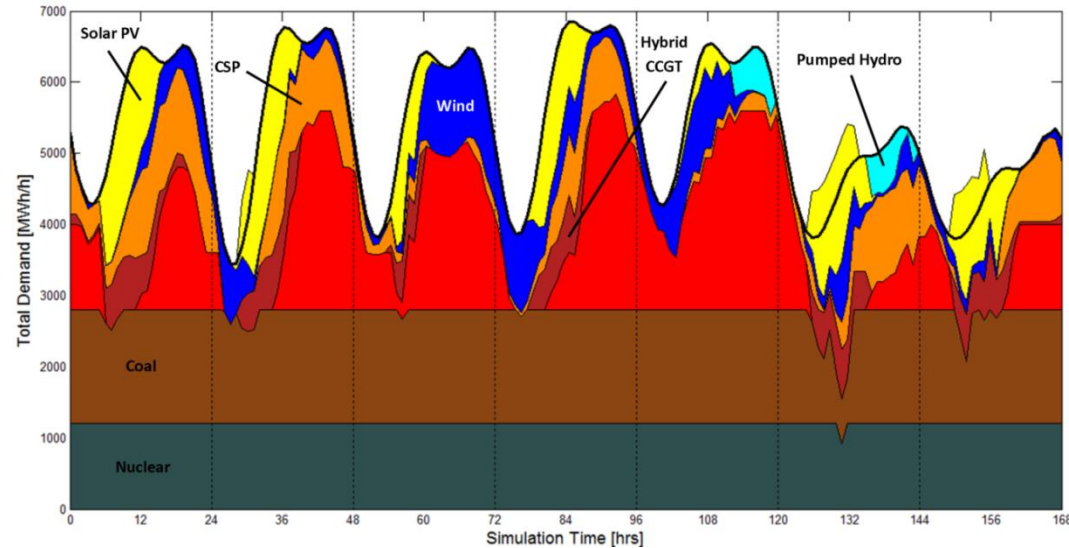
Intermittent renewables

Chase the best turbine efficiencies!

Effective combined cycle power plants

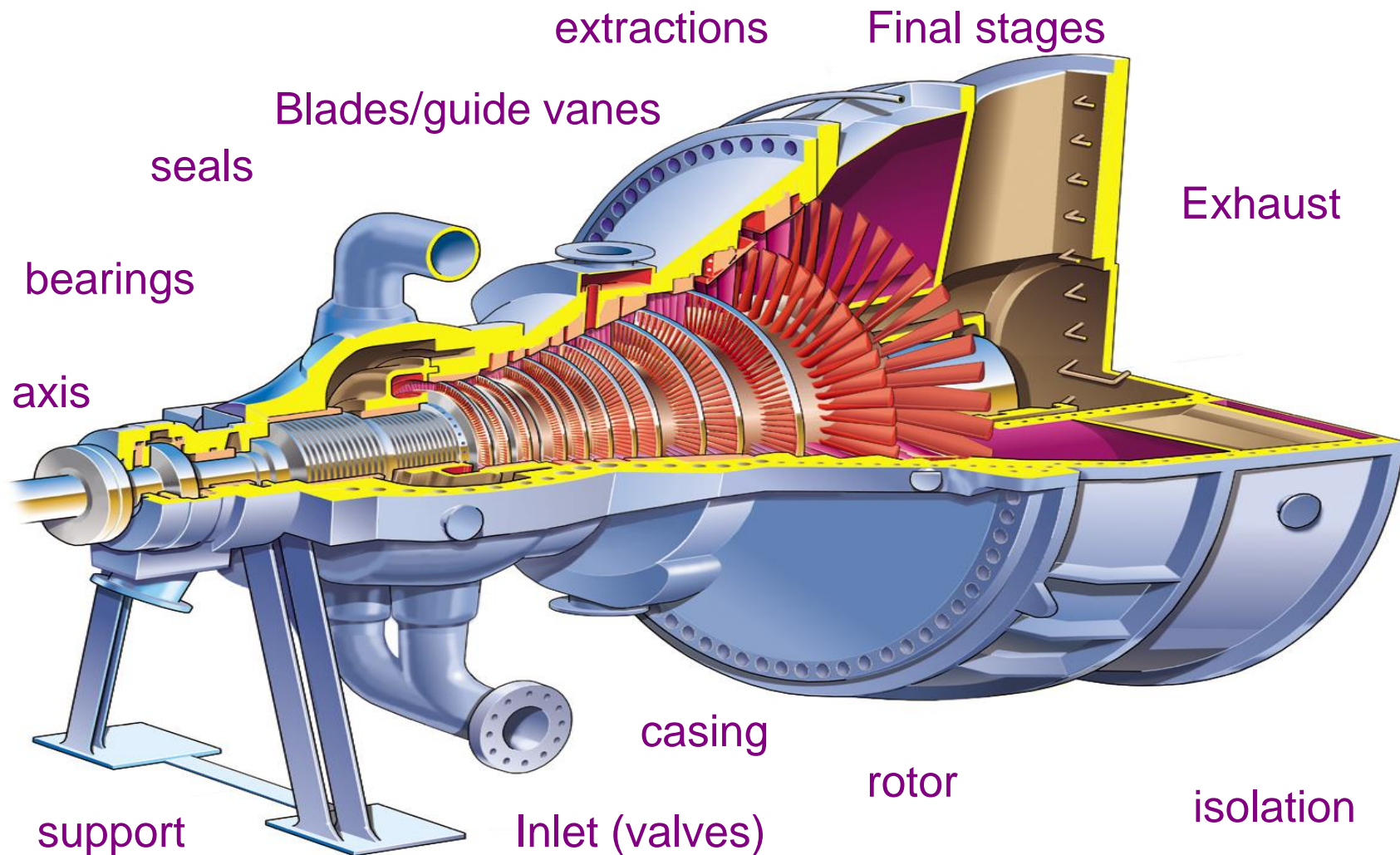
Highest efficiencies with district heating

Turbines for concentrating solar power

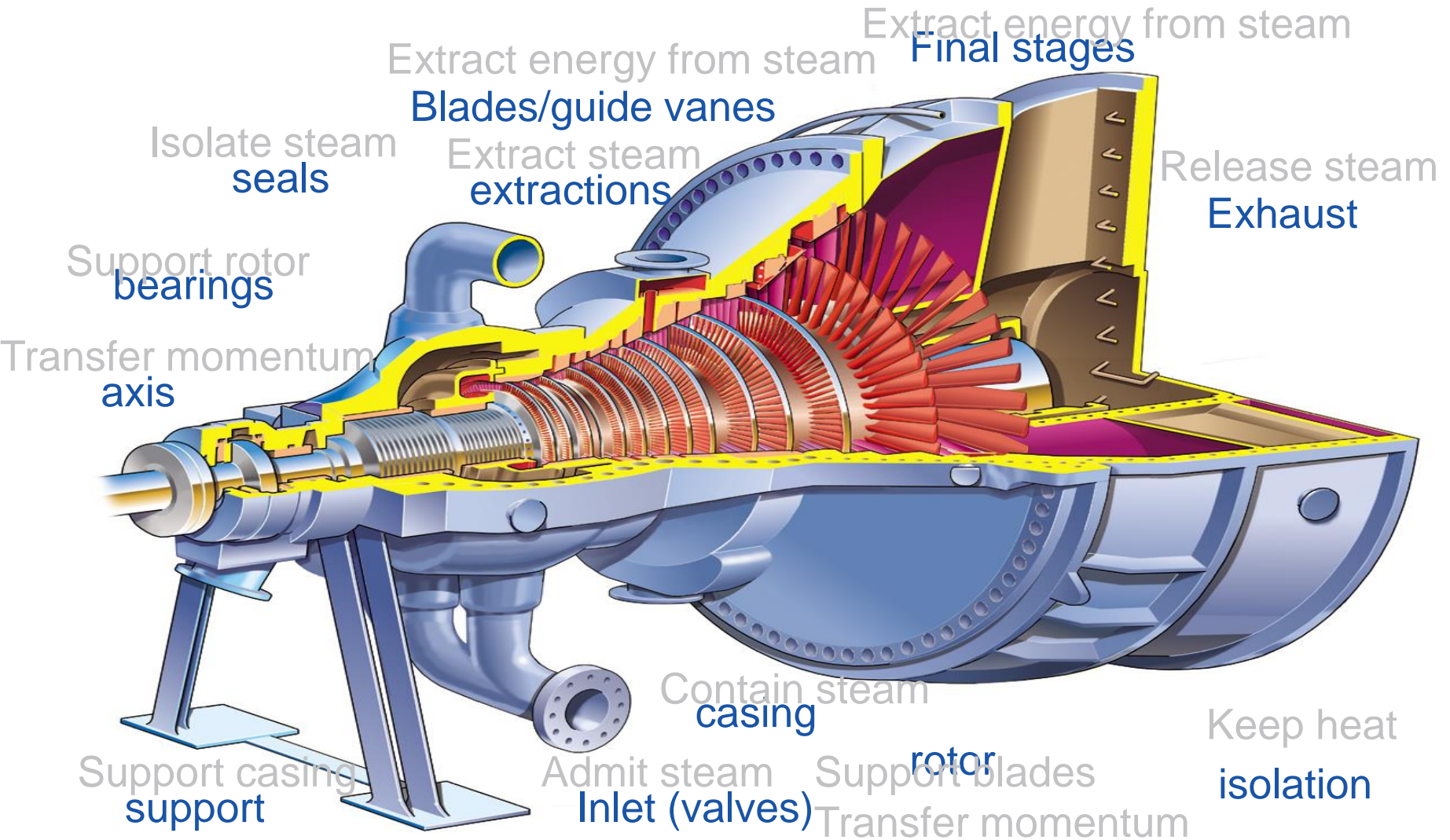


Gemasolar CSP plant

What are the main parts of a steam turbine?

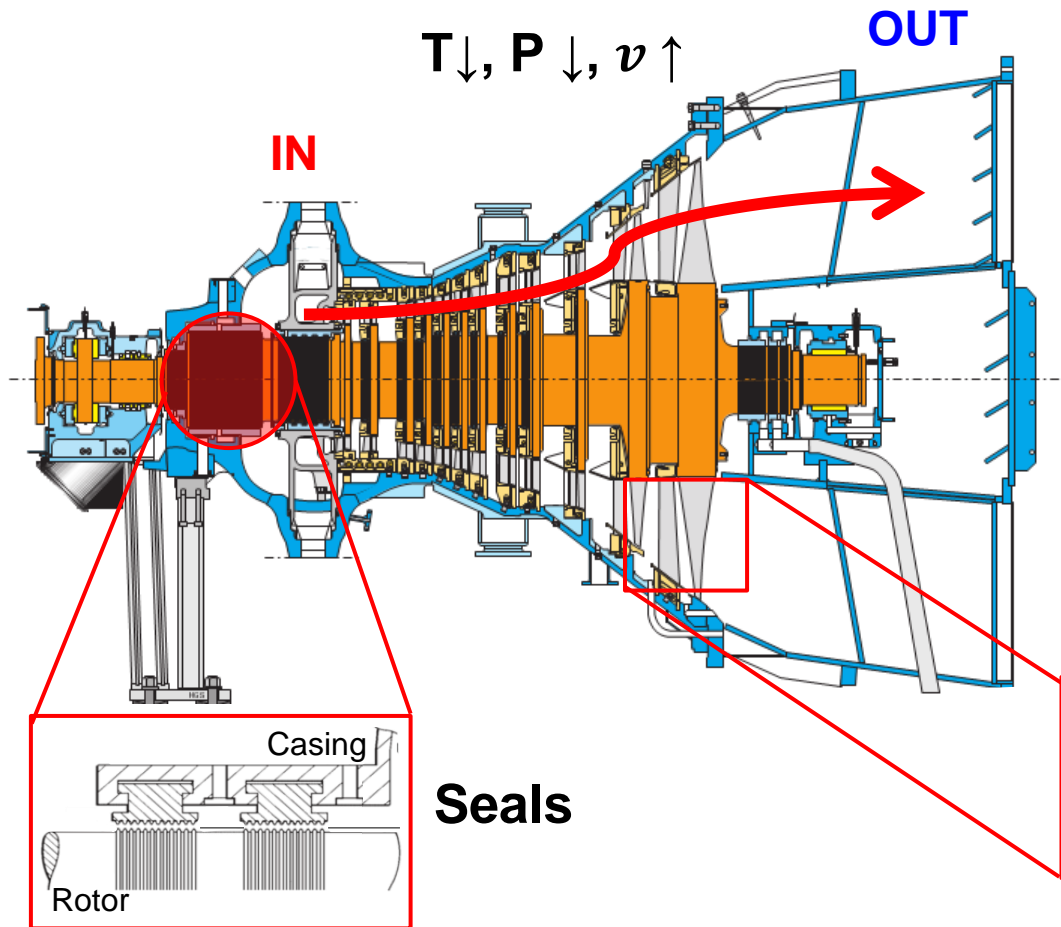


Main Function of these parts?



Steam Turbine – Overview

- Extract the thermal energy from steam,
- Shaft work → drive a generator.

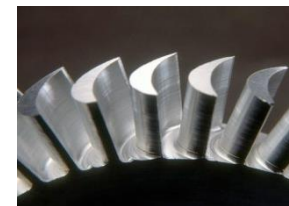


Steam Expansion

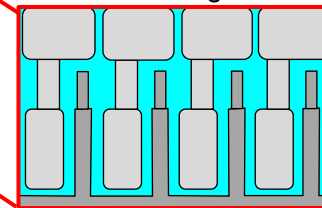
Rotor

Casing

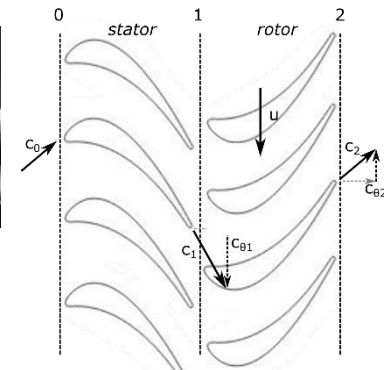
Blade rows



Casing



Rotor





Today's Lecture:

1. Blading
2. Control stages
3. Final stages
4. Seal technology
5. Inlets
6. Casings
7. Flexibility

What are major design objectives?

Today's Lecture:

- | | |
|--------------------|------------------------------|
| 1. Blading | efficiency, integrity, costs |
| 2. Control stages | efficiency, part load |
| 3. Final stages | efficiency, integrity |
| 4. Seal technology | performance, integrity |
| 5. Inlets | low losses, costs |
| 6. Casings | thickness, tightness, costs |
| 7. Flexibility | fast response, lifetime |

What are major design objectives?

**performance, reliability,
flexibility, costs**

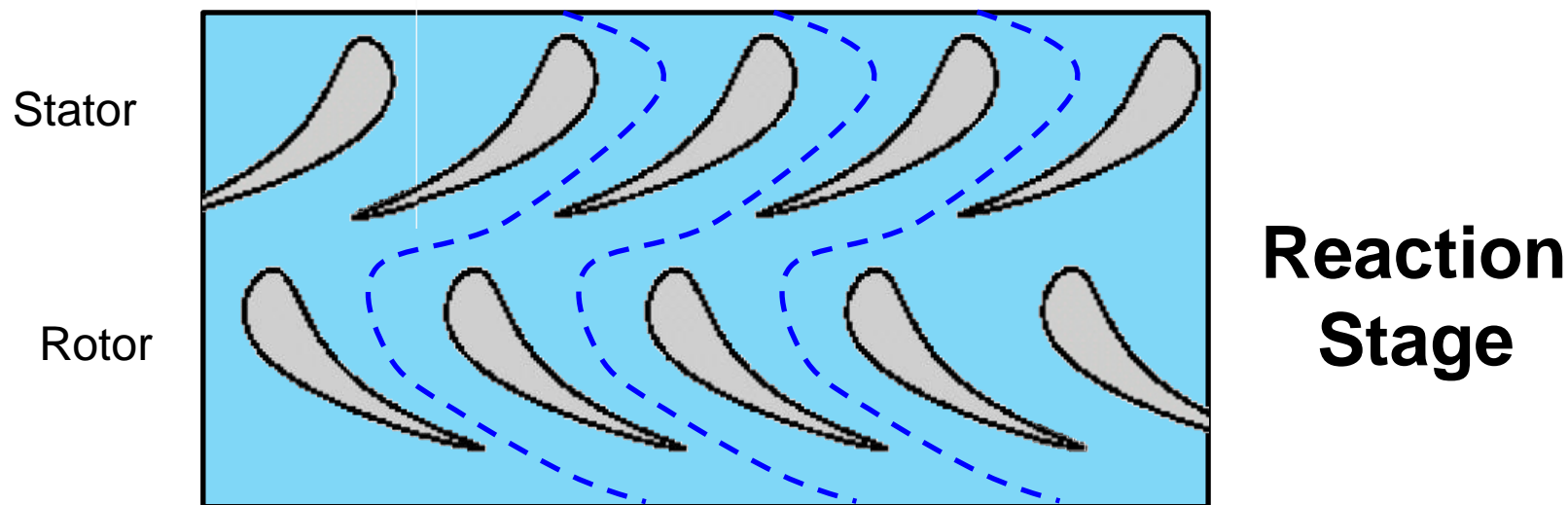
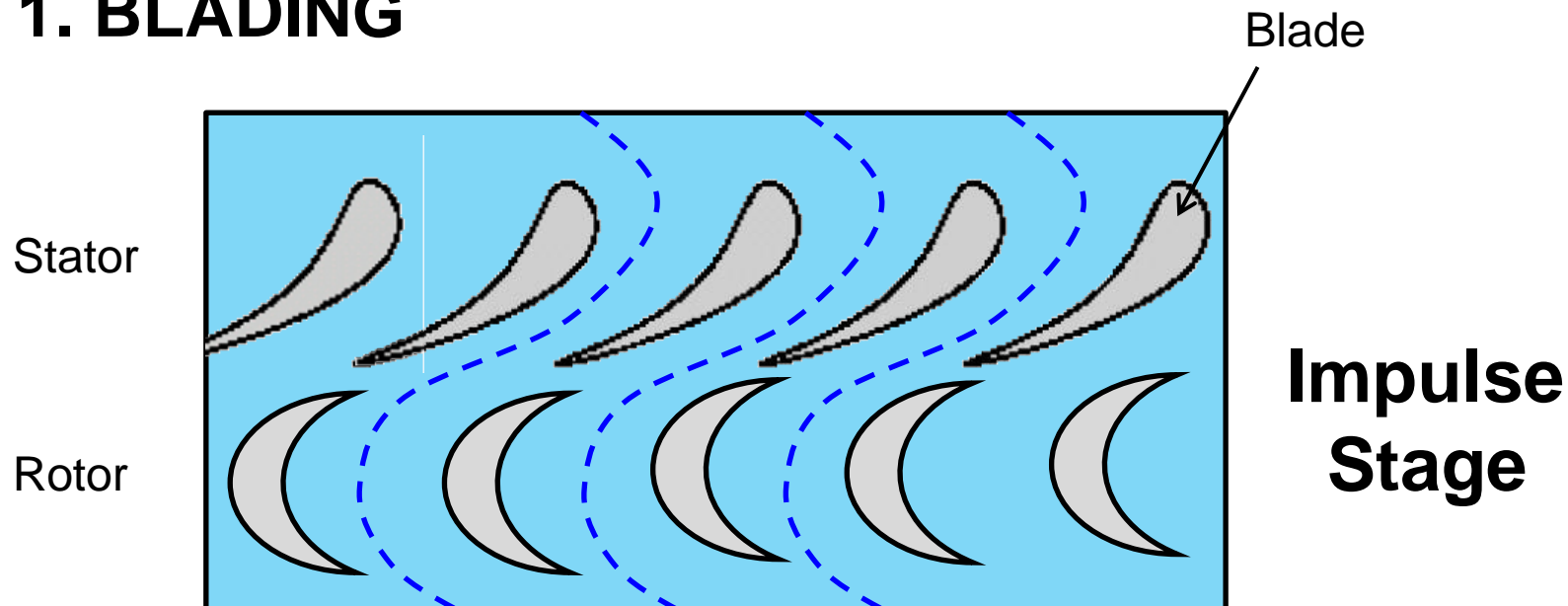
1. BLADING

- Main function: Extract steam energy
 - High temperature and pressure at the inlet
 - Take that energy into shaft work

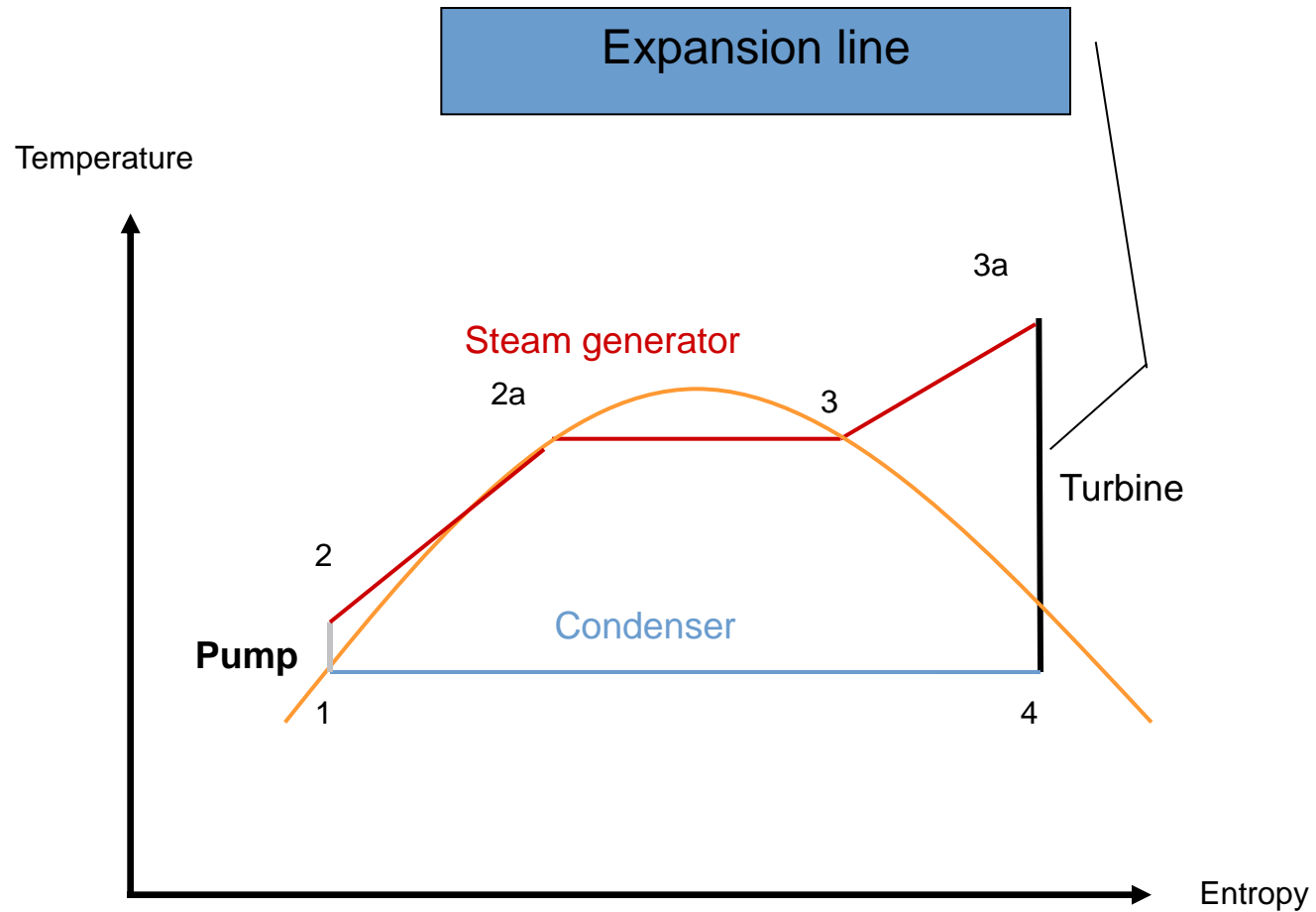
Think of a hydro turbine !

- Main principles of a steam turbine
 - Rotation
 - Deviation of the flow
- Blades deviate the flow
 - Impulse (low reaction)
 - Reaction

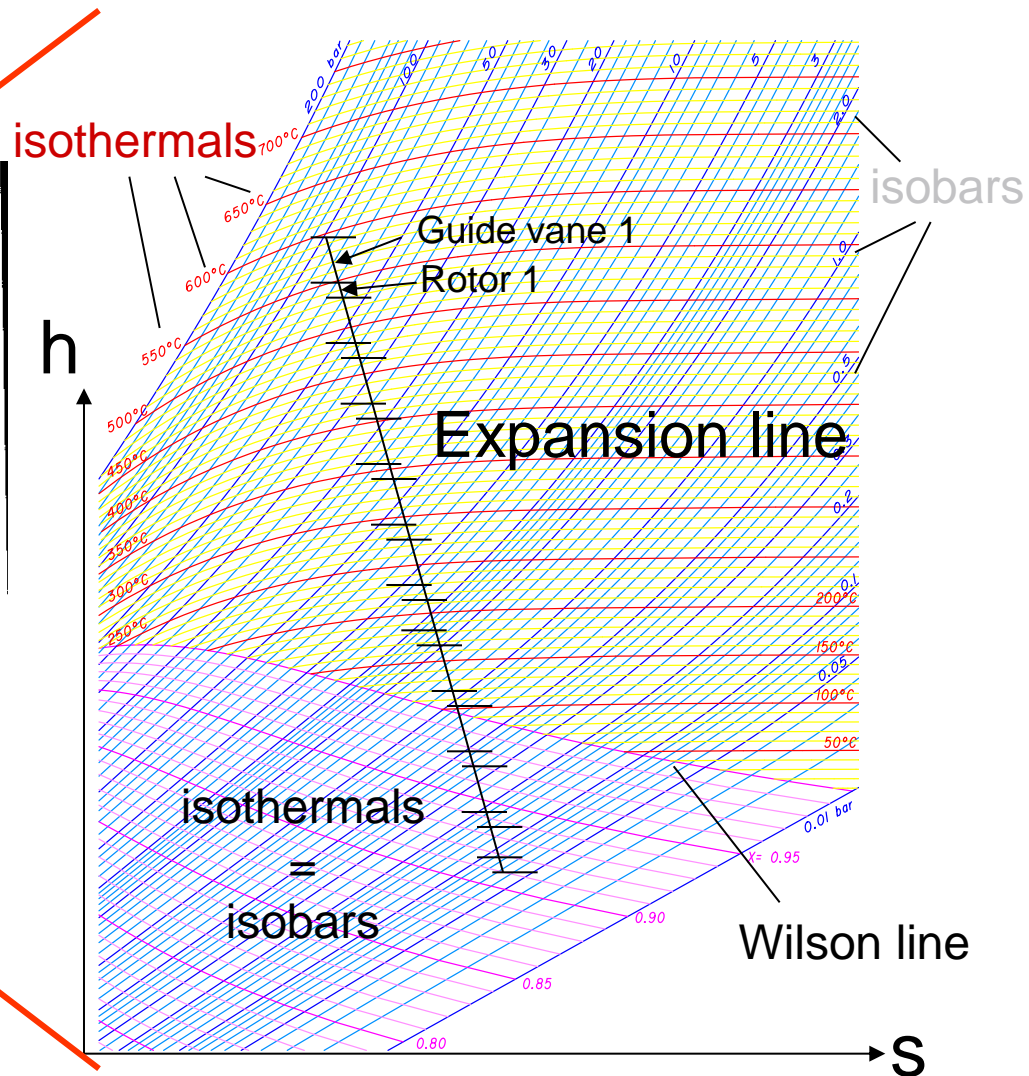
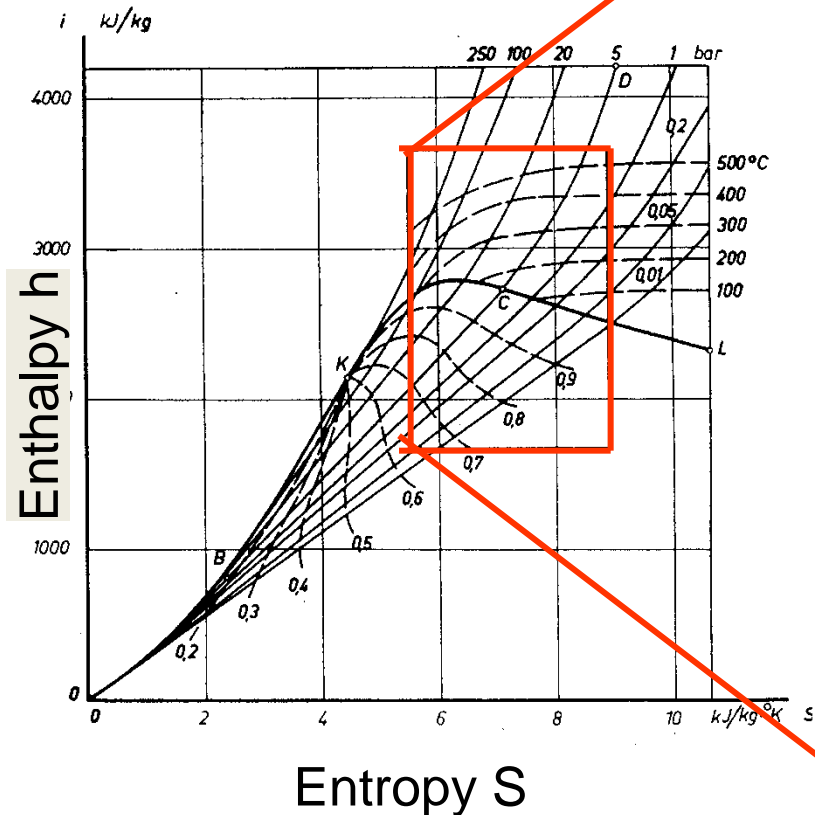
1. BLADING



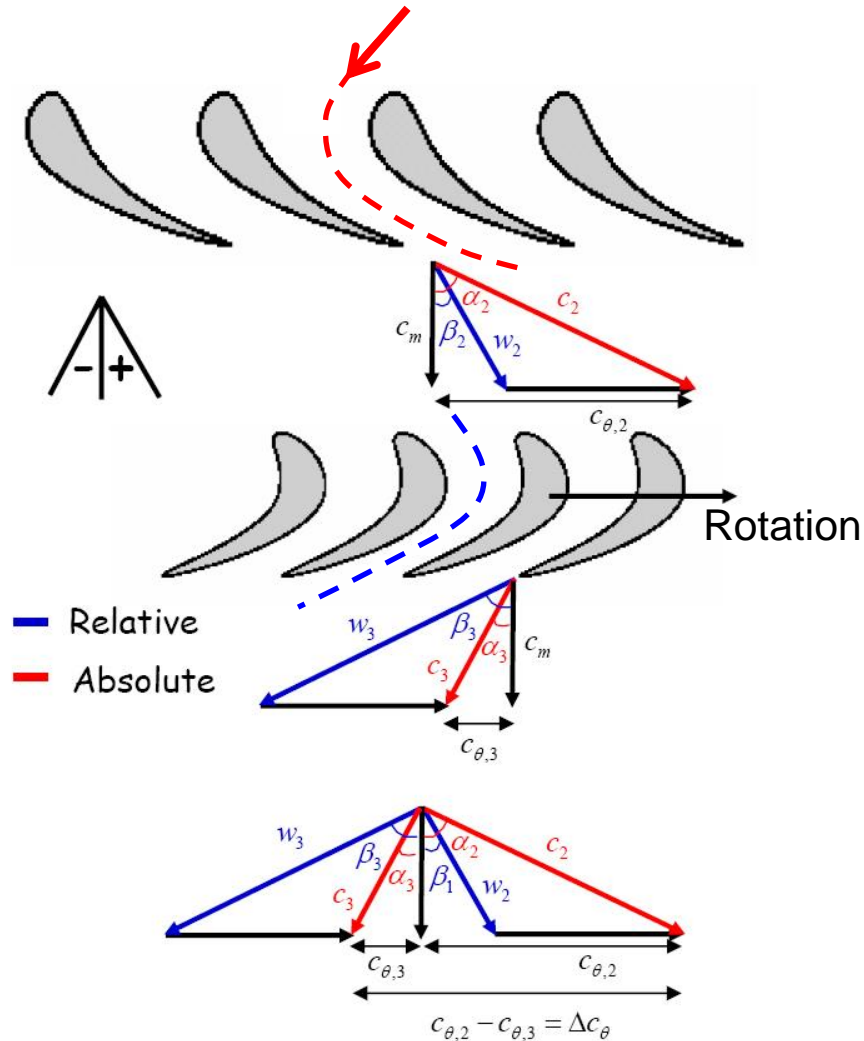
Rankine Cycle



Mollier Diagram



Velocity triangles

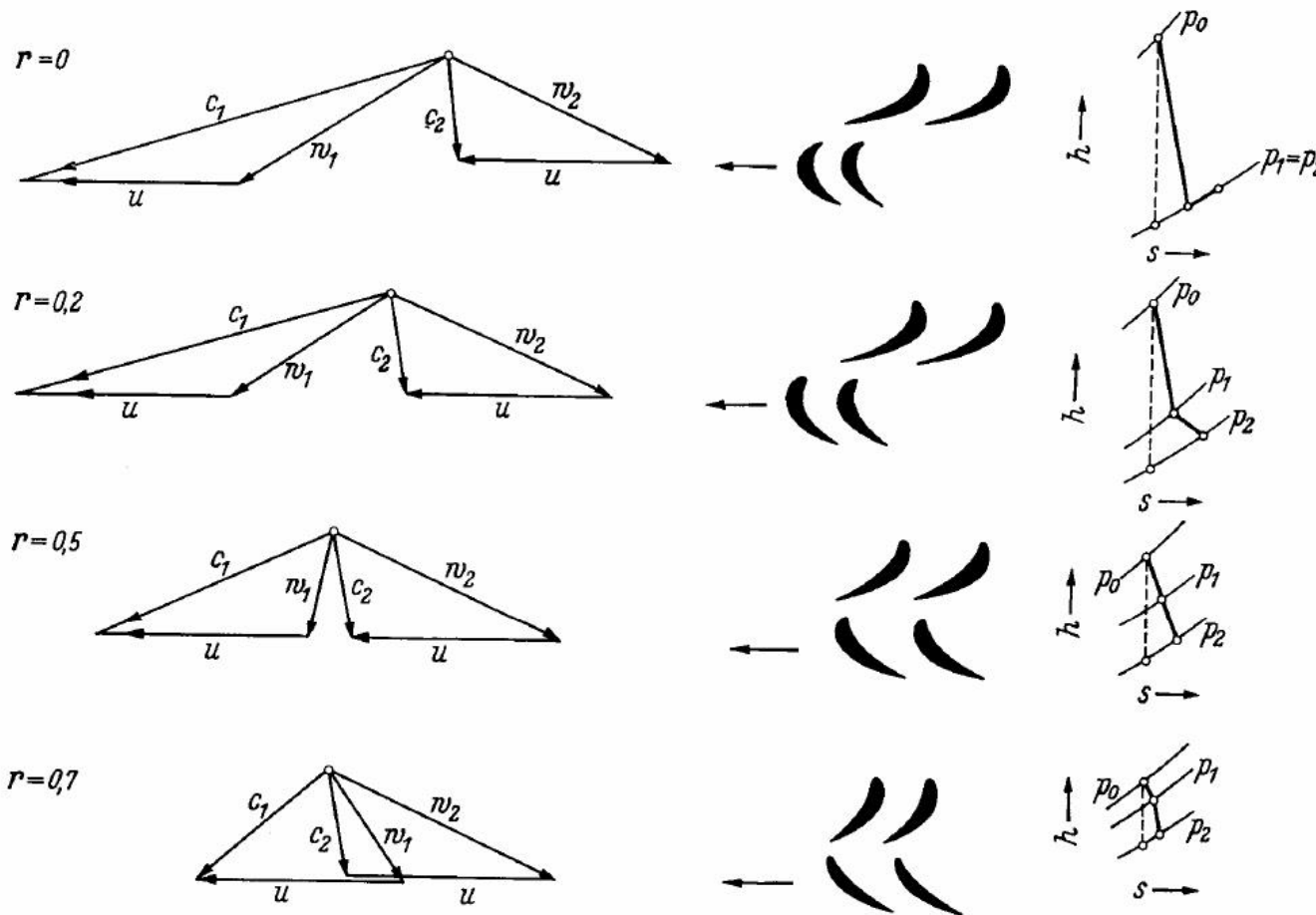


Stator row

Rotor row

$$\vec{C}_i = \vec{W}_i + \vec{U}$$

Reaction



$$R = \frac{\Delta h_{is} \text{ Rotor}}{\Delta h_{is} \text{ Stage}}$$

$$R \sim \frac{\Delta p \text{ Rotor}}{\Delta p \text{ Stage}}$$

"Traupel"

Euler's turbine equation

- Conservation Principles
 - Mass
 - Momentum
 - Energy

$$w = \Delta h = u \times \Delta c_{\theta}$$

w	specific work	[J/kg] [W/(kg/s)]
h	enthalpy	[J/kg]
u	rotor blade velocity	[m/s]
c_{θ}	absolute swirl velocity	[m/s]

Smith diagram

Stage loading coefficient:

$$\psi = \frac{\Delta h_0}{u^2}$$

Flow coefficient:

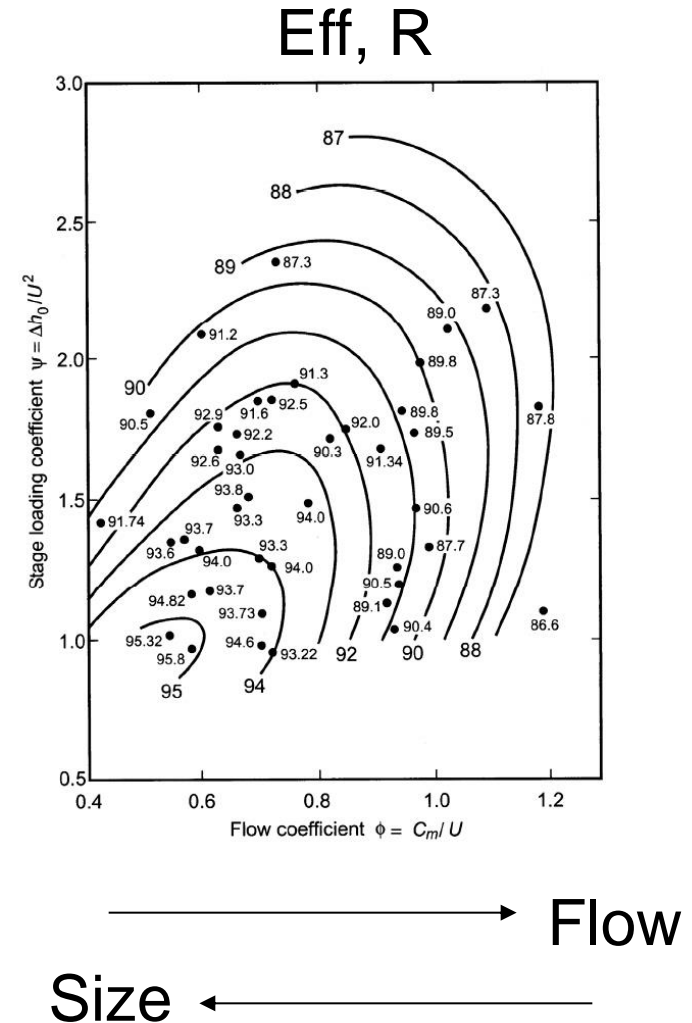
$$\phi = \frac{c_m}{u}$$

U: rotor velocity

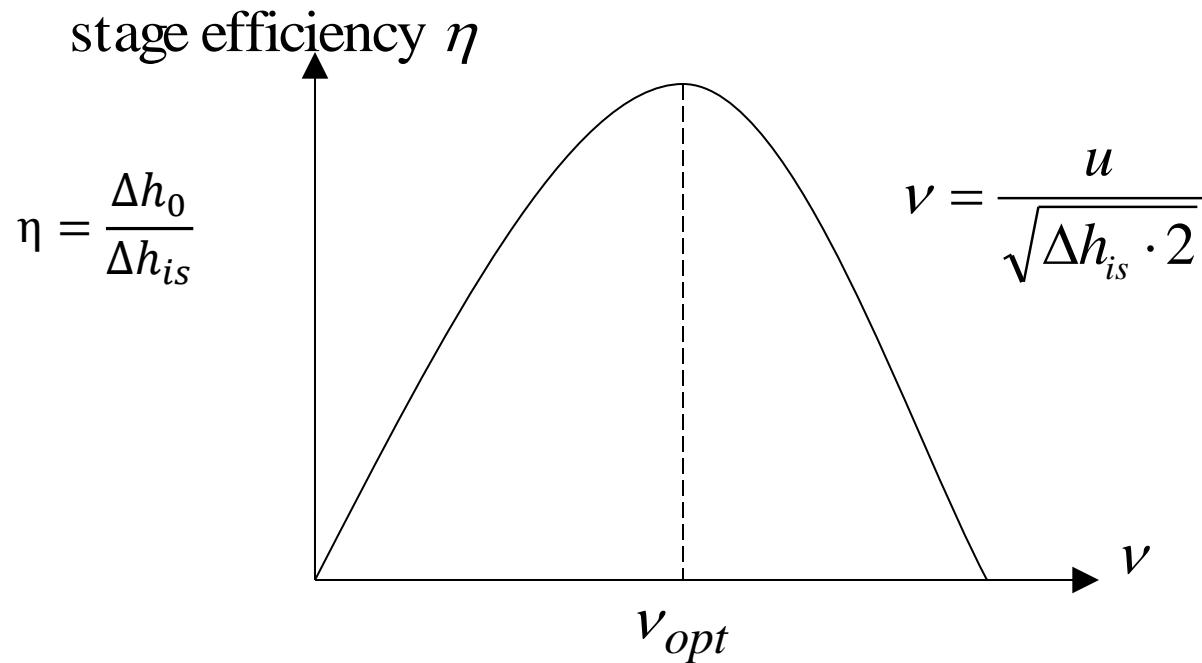
C_m: axial flow velocity

Aero.
loading

Number
of stages



Velocity number



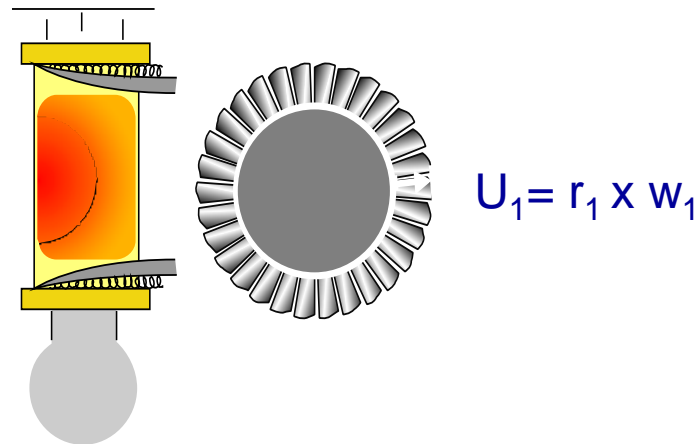
u = rotor blade velocity $[m/s]$

Δh_{is} = isentropic enthalpy drop $[J/kg]$

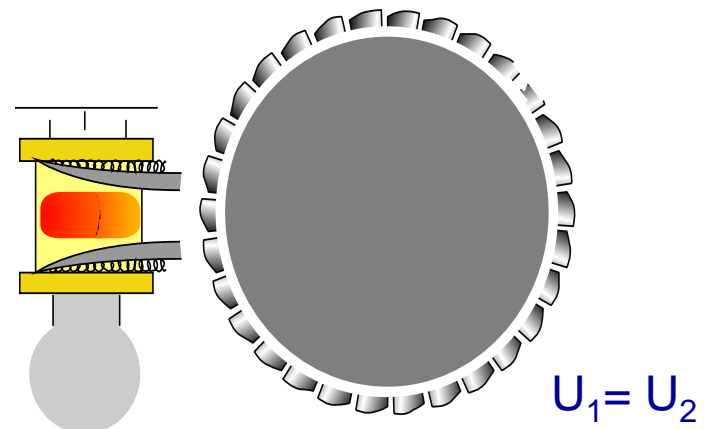
BLADING – rotor speed

- High specific rotor speed enables use of long blades for improved efficiency
- Small tip leakage area due to small rotor diameter

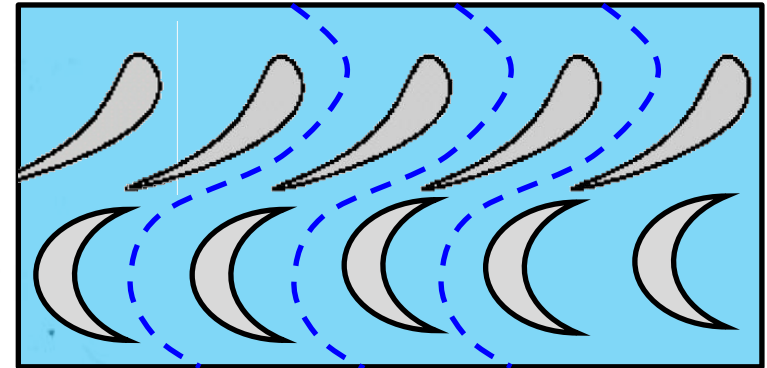
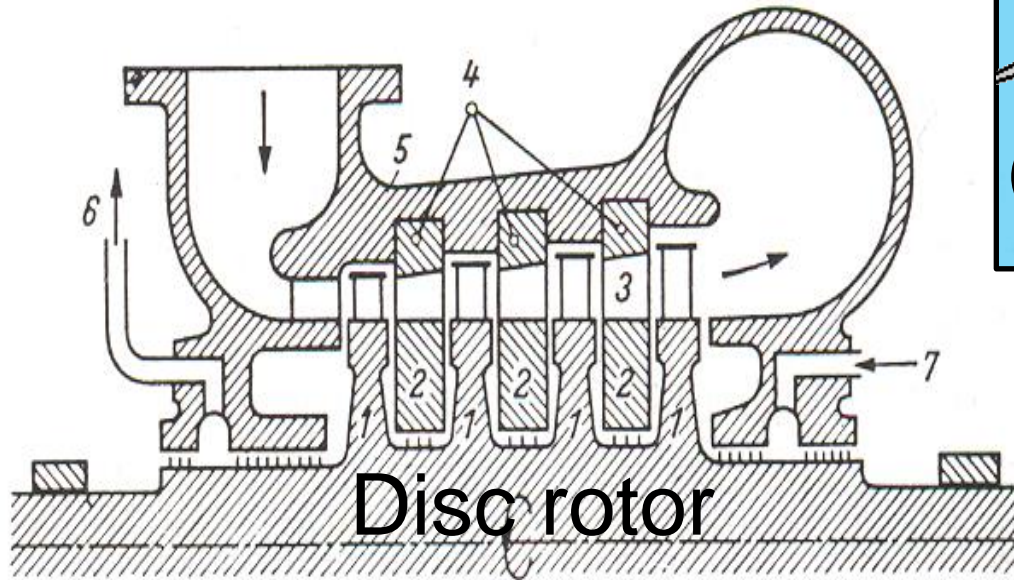
SST700 HP turbine



Traditional turbine



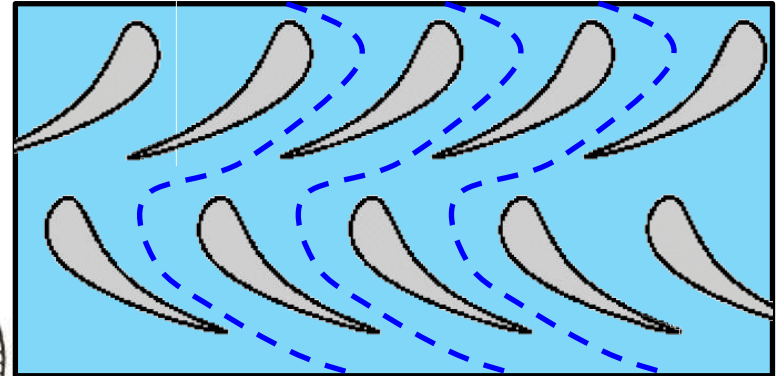
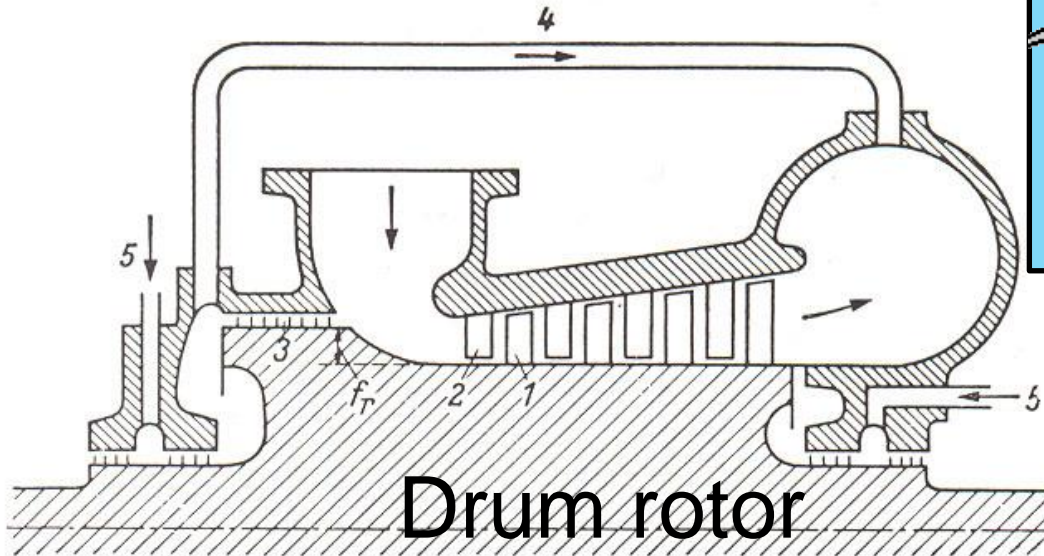
1. BLADING



Impulse blading

- Nearly no reaction
- Small or no pressure drop over the rotor
- High work output/stage (less stages)
- Diaphragm design
- Less leakage losses
- More axial length/stage

1. BLADING



Reaction blading

Reaction about 0.5

Pressure drop over rotor requires
balance piston

Less work output /stage (more stages)

Better aerodynamics due to less turning

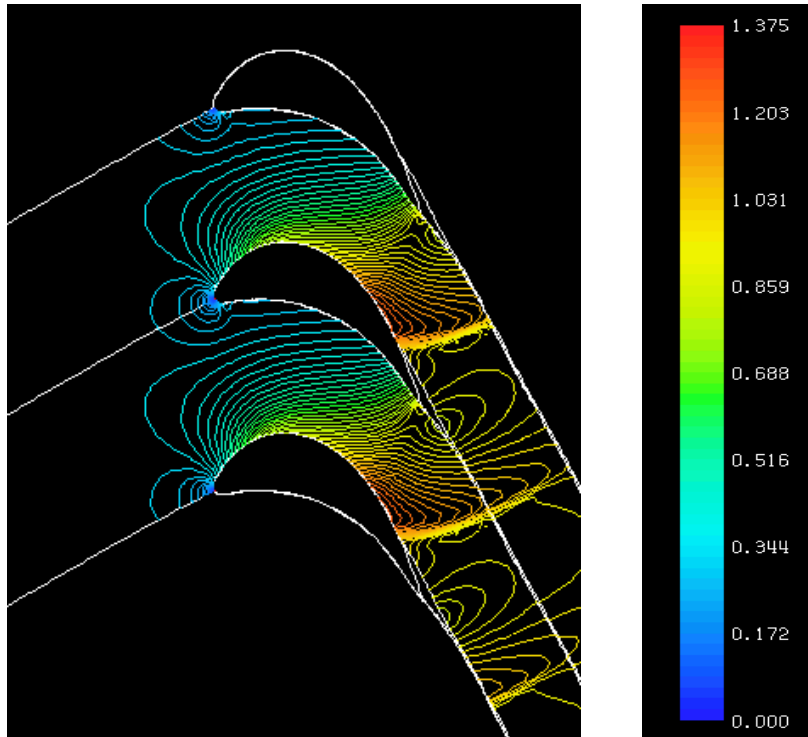
1. BLADING

Design aspects:

- Aerodynamics
- Integrity
- Roots
- Shroud Seals
- Manufacturing and assembly



Profile losses



Mach Number in a rotor blading - transonic

Friction losses

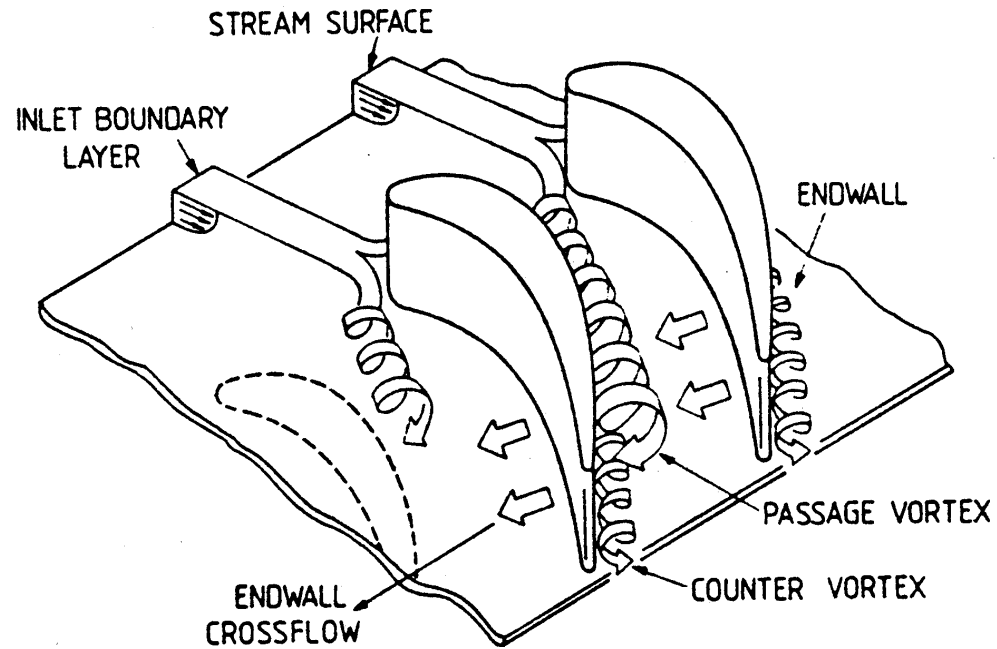
Trailing edge losses

Shock losses

Separation losses

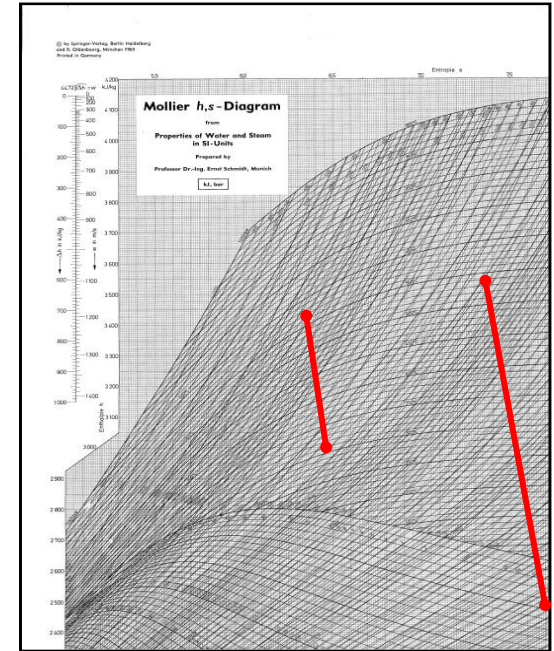
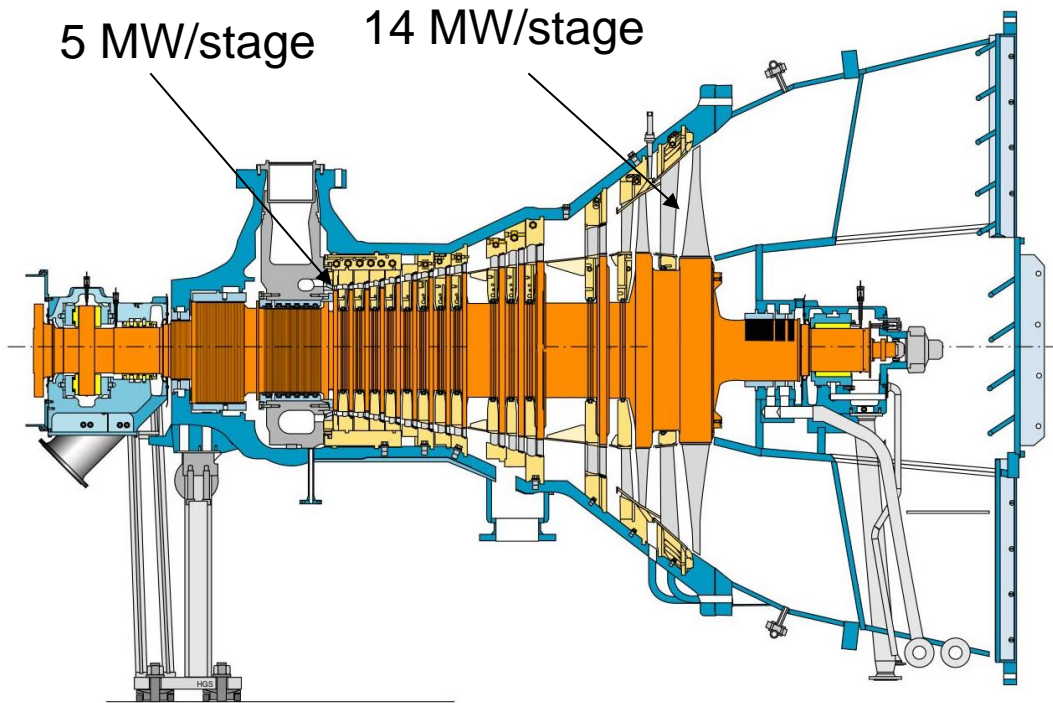
$$Ma = \frac{\text{velocity}}{\text{sound velocity}}$$

Secondary flow



Endwall flow models by Klein and Langston

1. BLADING



Where are the challenges?

Pressure ratio : 2000
Volume ratio: 700

1. BLADING

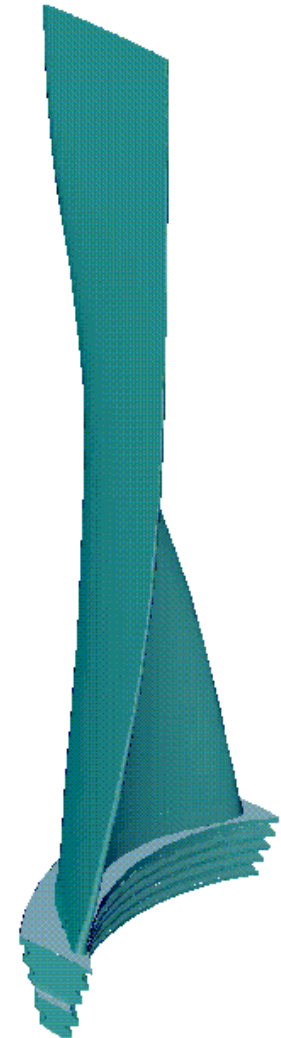
Large centrifugal loads (static tension)	LP
Bending/torsion loads	LP
Large thermal loads (LCF, creep)	HP, IP
Unsteady steam loads (HCF)	HP, IP, LP
Stress concentration in notches	HP, IP, LP
Corrosion	LP
Erosion by particles	LP
Oxidation	IP, LP

1. BLADING

Reduce thermodynamic losses!

- Profile losses
 - Secondary losses
 - Leakage losses
 - Axial gap losses
 - Moisture losses
 - Exhaust losses
 - Unnecessary losses

1. BLADING – typical blade data



			HP	LP
Bladelength	l	[mm]	34	866
Length/Chord	L/C	[-]	1.1	4.9
Diameter ratio	D_y/D_i	[-]	1.2	2.2
Blade velocity	u	m/s	150	450
Reynoldsnumber	Re	$[-] \cdot 10^5$	40	4
Machumber	Ma	[-]	0.2	1.3

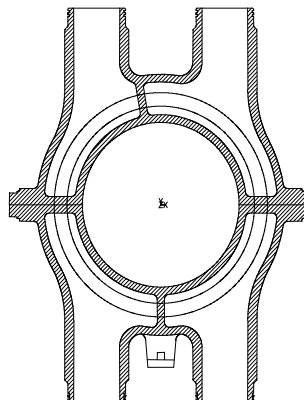
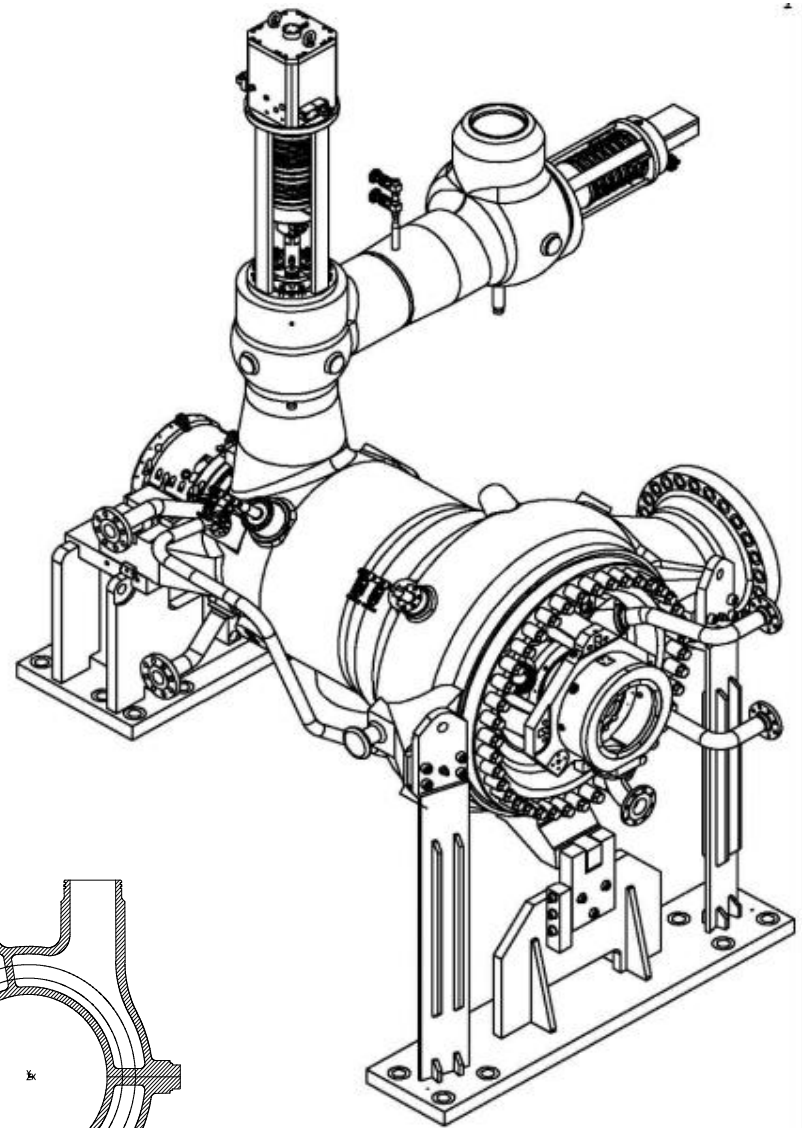
2. CONTROL STAGE

Partial admission
to control load

Several admission valves

Require zero reaction stage

High dynamic loads



Partial admission nozzle box

HP/turbine with control stage

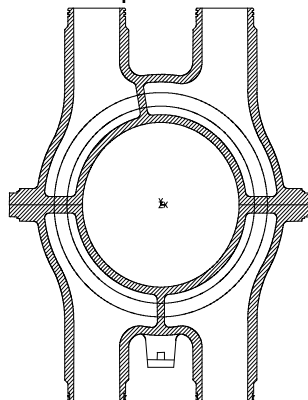
2. CONTROL STAGE

Partial admission
to control load

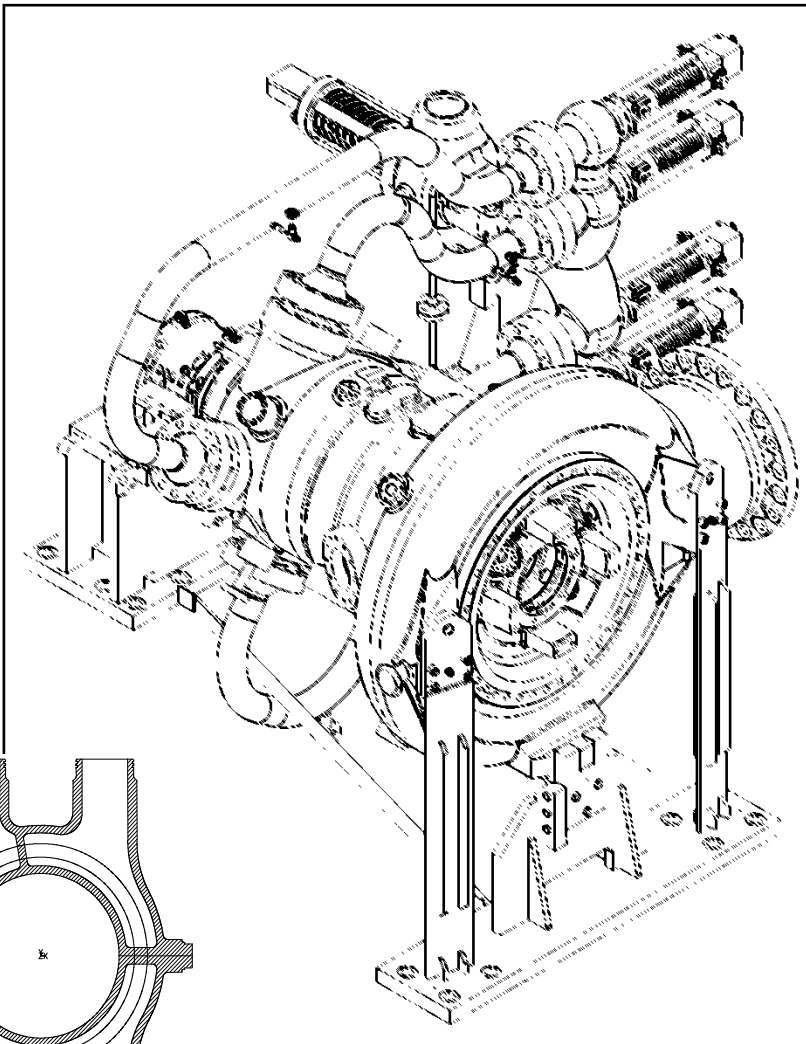
Several admission valves

Require zero reaction stage

High dynamic loads

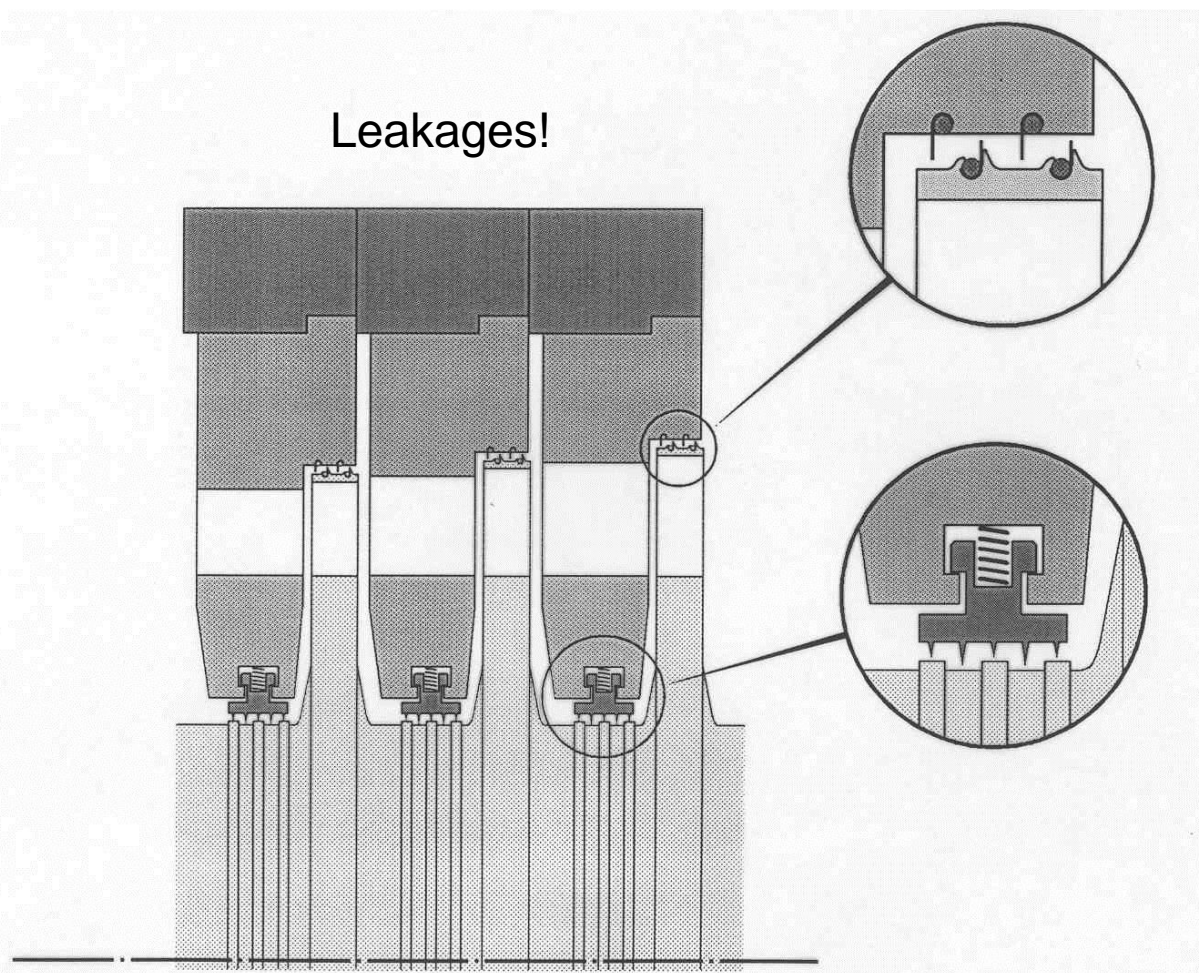
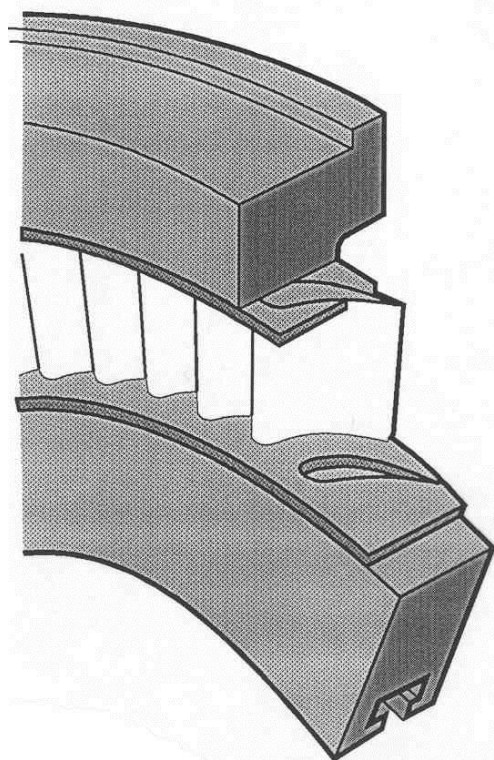


Partial admission nozzle box

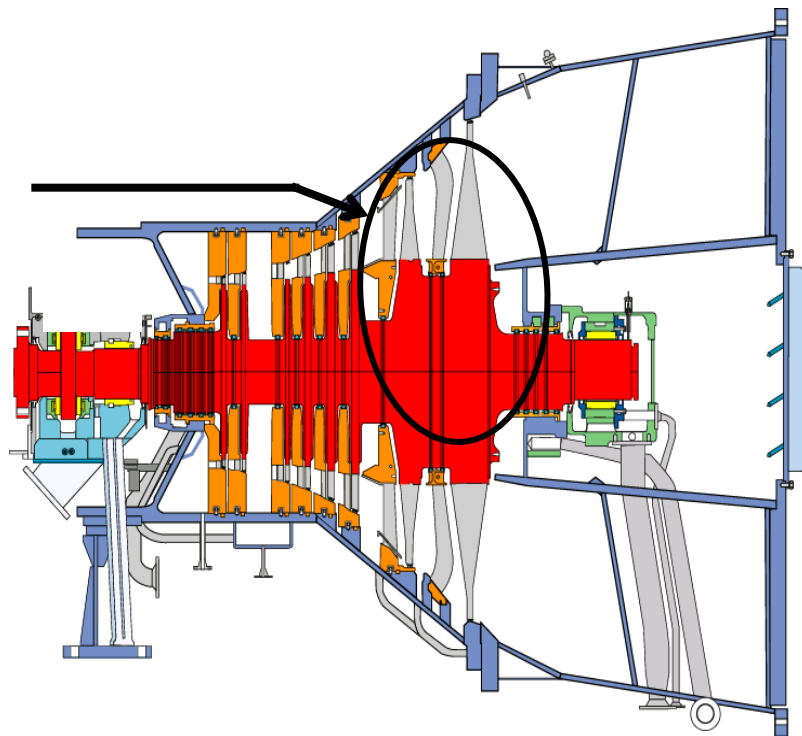


HP/turbine with control stage

Diaphragms principle design

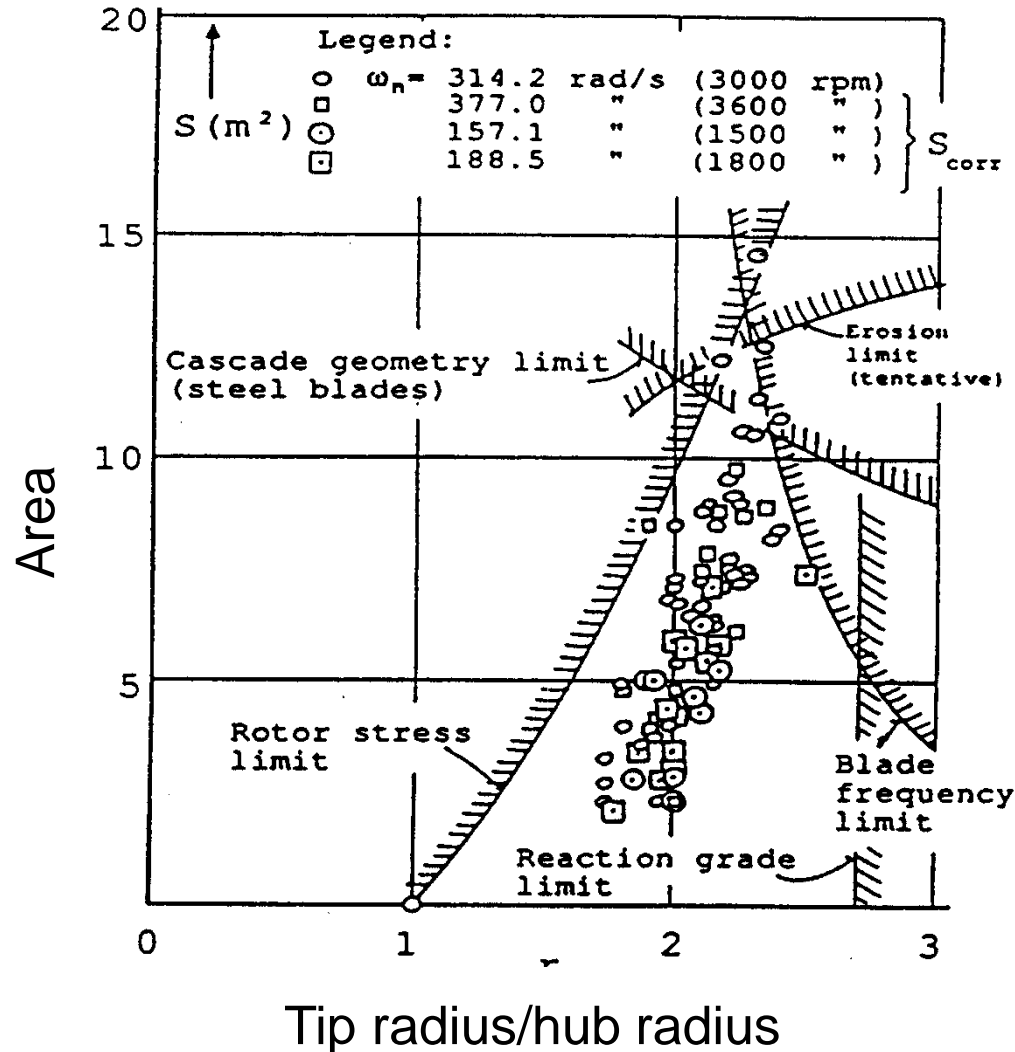


3. FINAL STAGES



Where are the challenges?

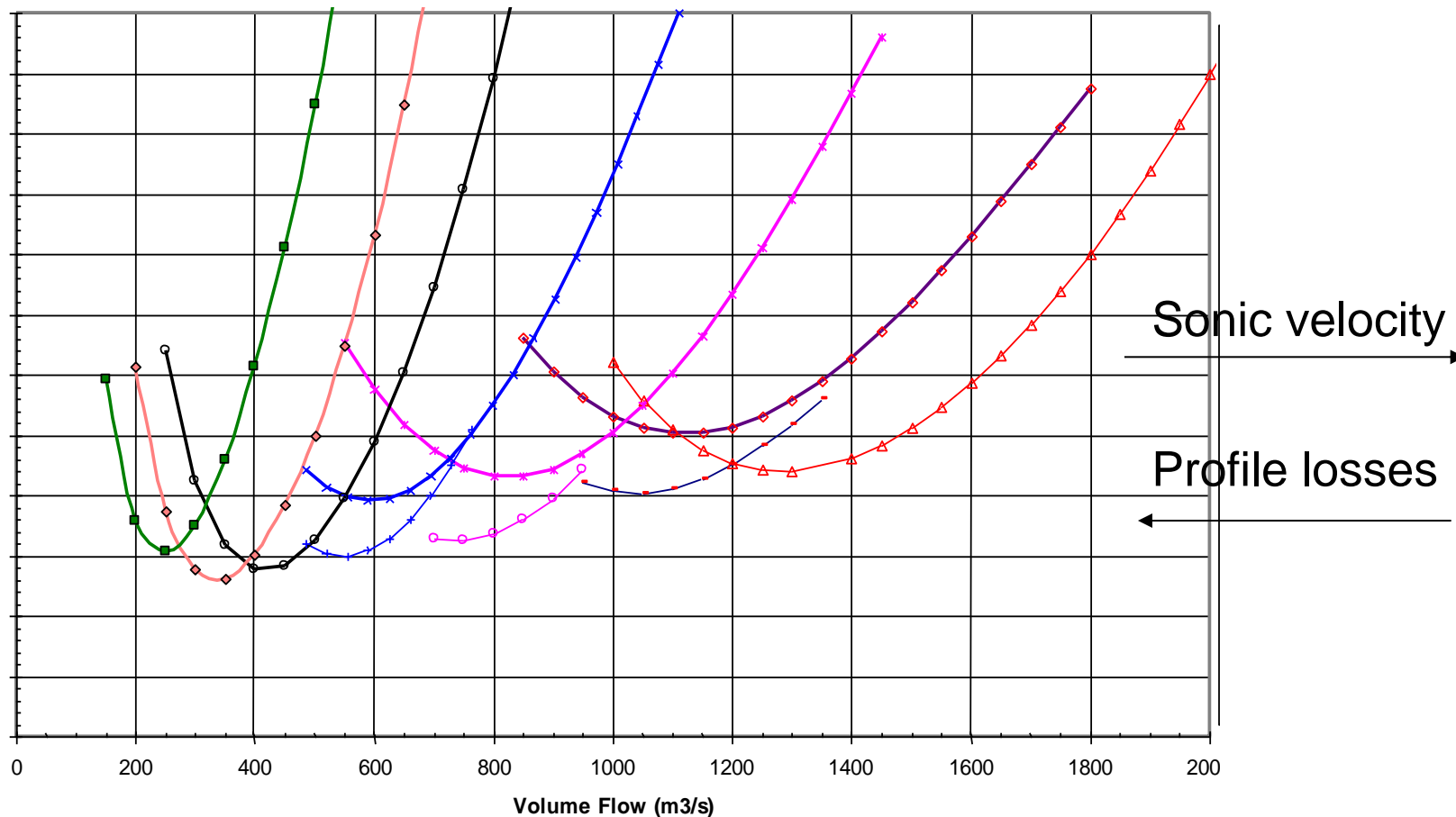
3. FINAL STAGES – limits



Rotor stress
Erosion
Cascade geometry
Blade frequency

"On the design limits of steam turbine stages"
Gyarmathi, Schlachter, 1988

3. FINAL STAGES – Exhaust losses



3. FINAL STAGES - roots

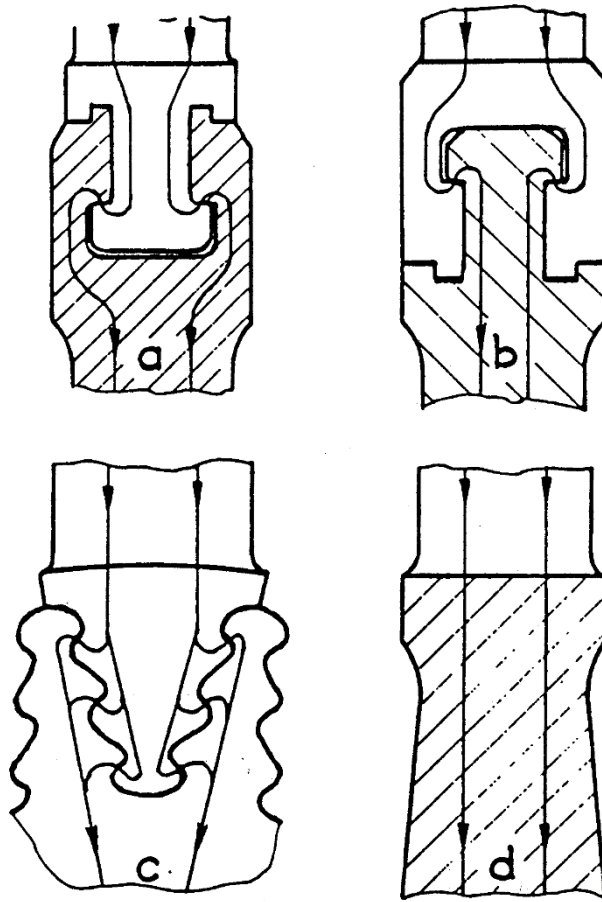


Figure 6. Load Transfer for Various Roots. a) Circumferential Internal Groove, b) Straddled Root [3]. c) Axial Sawtooth Root, and d) Blade Integral With Rotor [7].

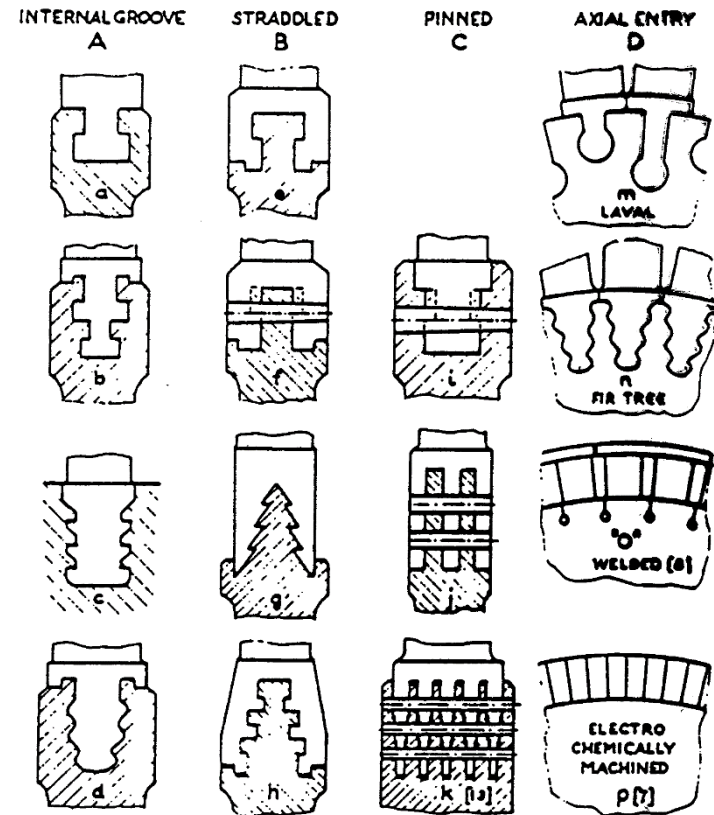
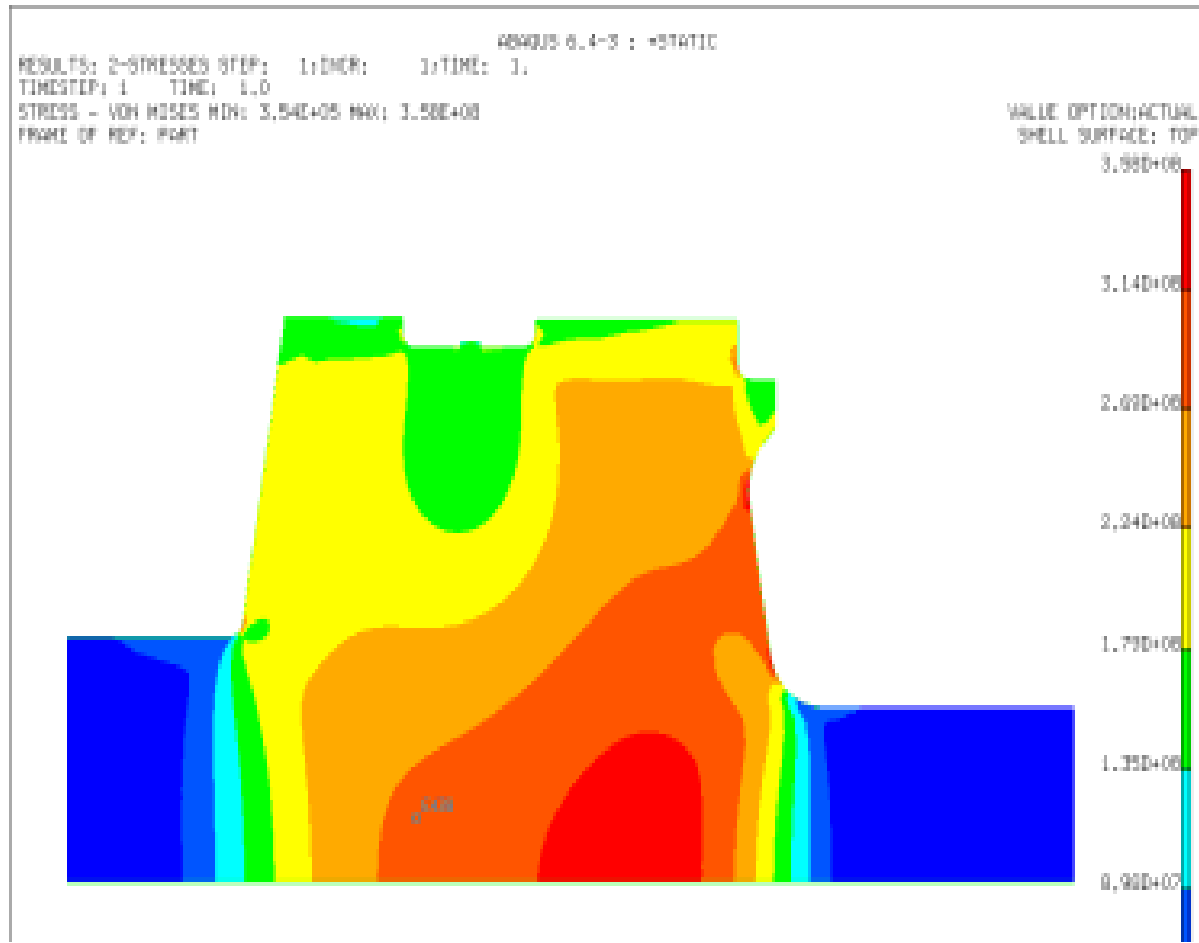
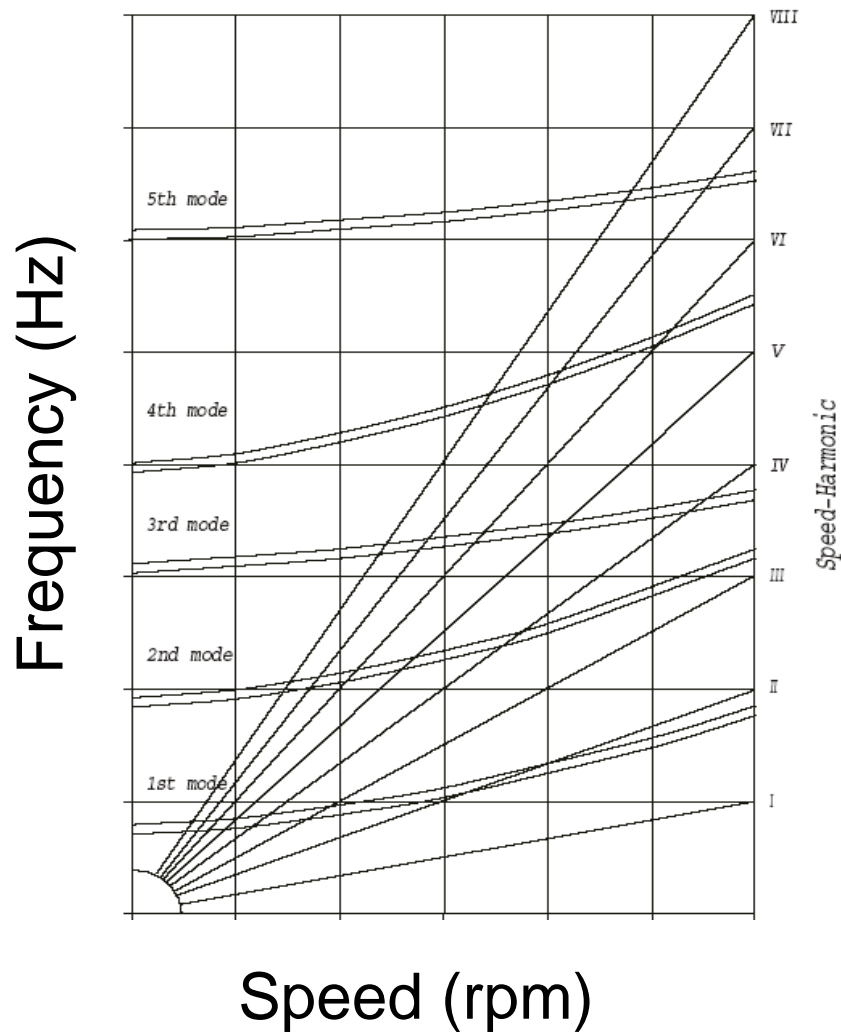


Figure 7. Blade Fastenings.

3. FINAL STAGES – example root stress analysis

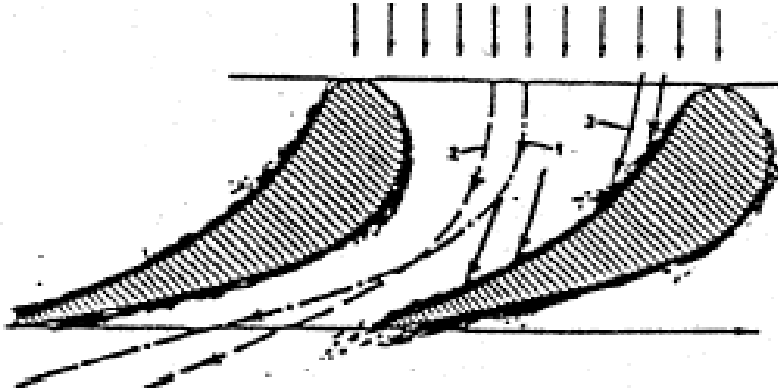


3. FINAL STAGES – blade vibration

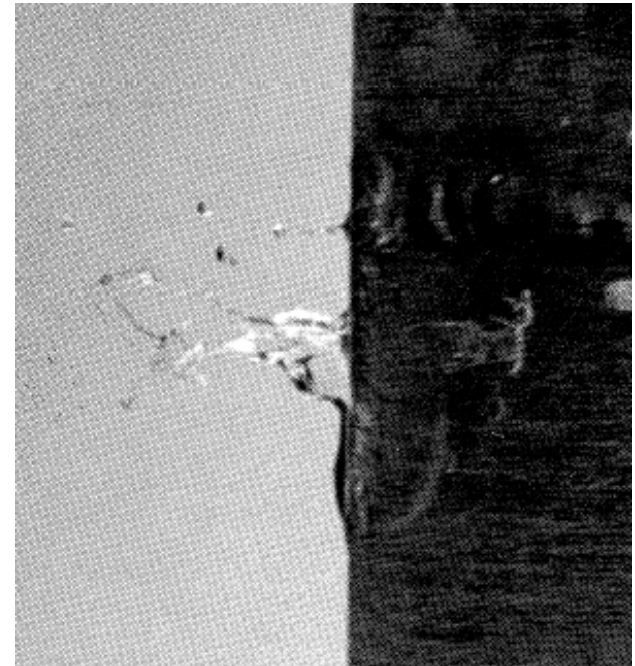
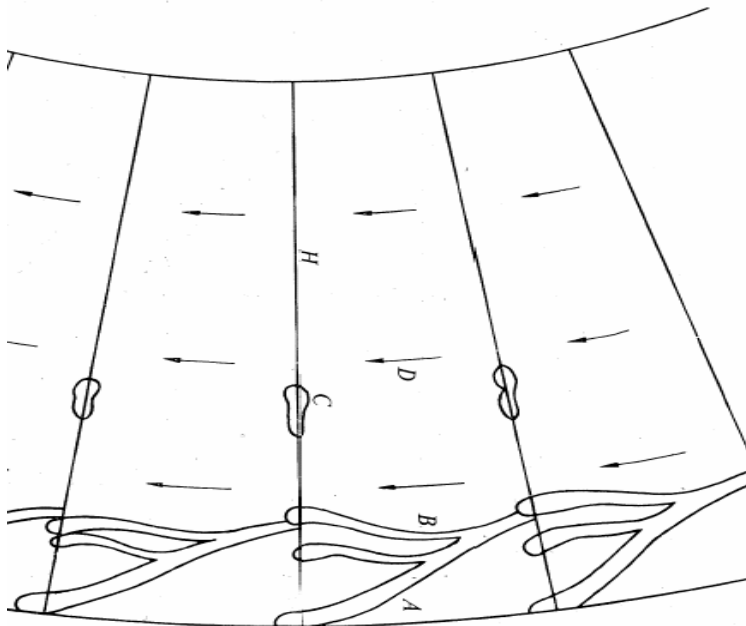
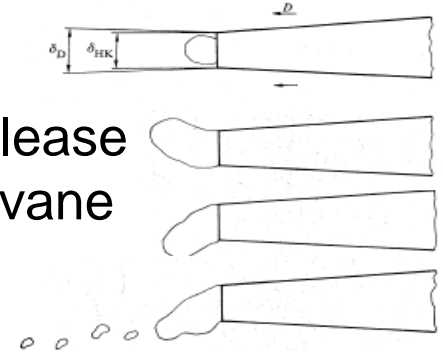


Campbell diagram

3. FINAL STAGES – moisture



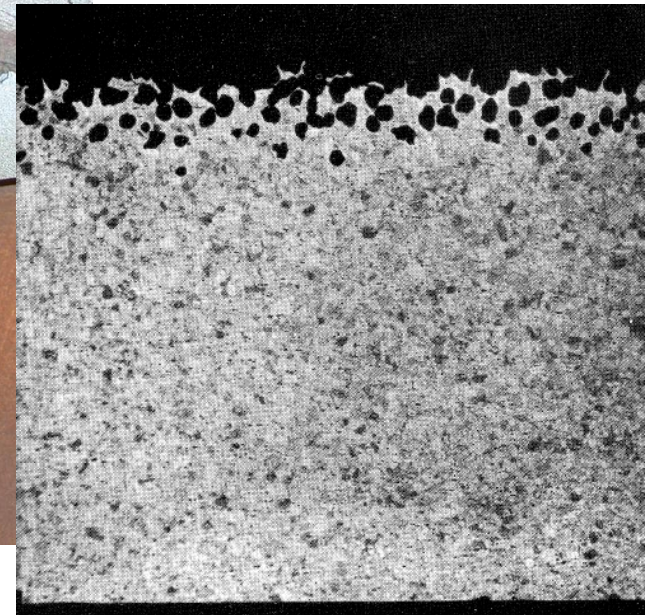
Collection and release of moisture from vane trailing edge



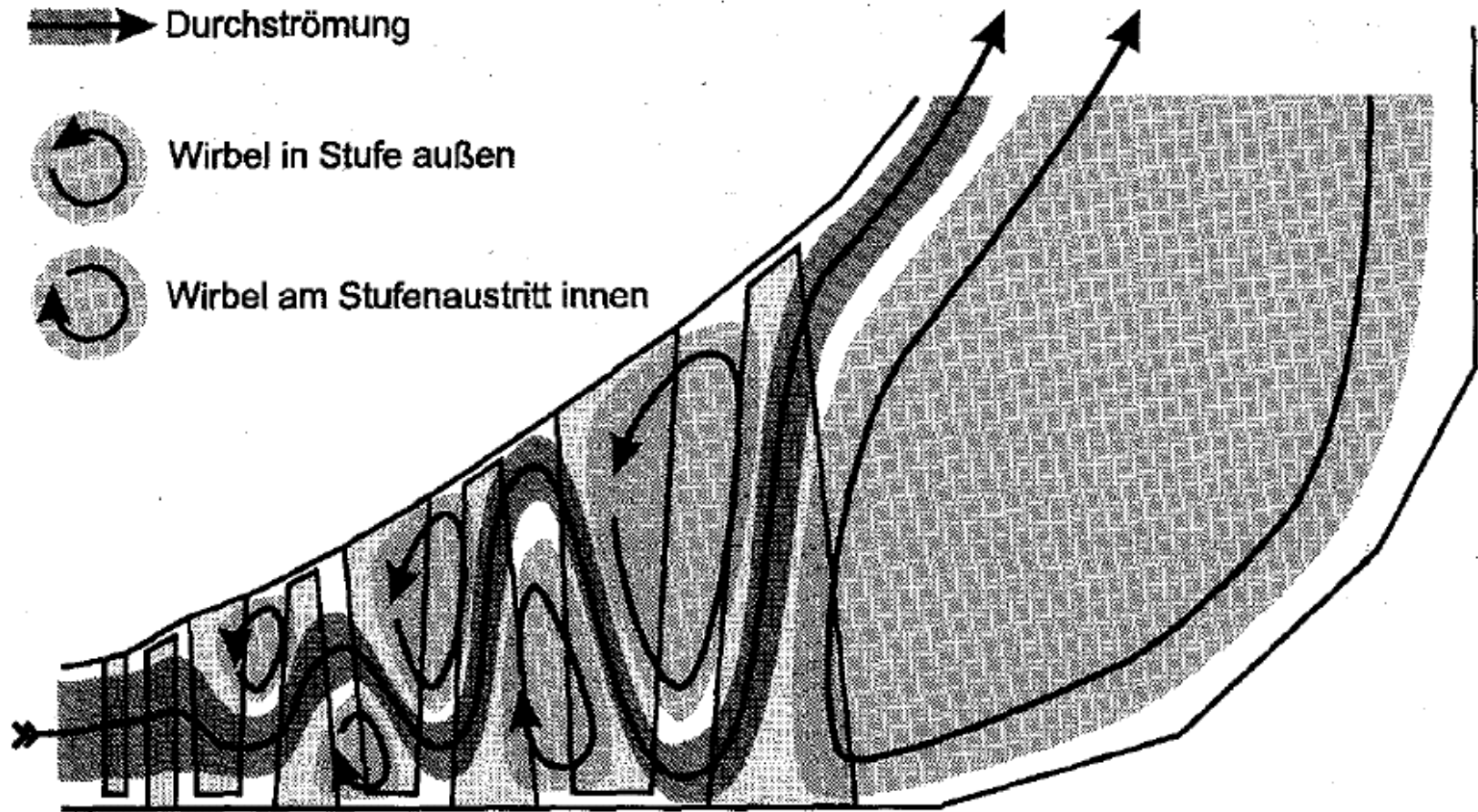
3. FINAL STAGES – Erosion damage on blades



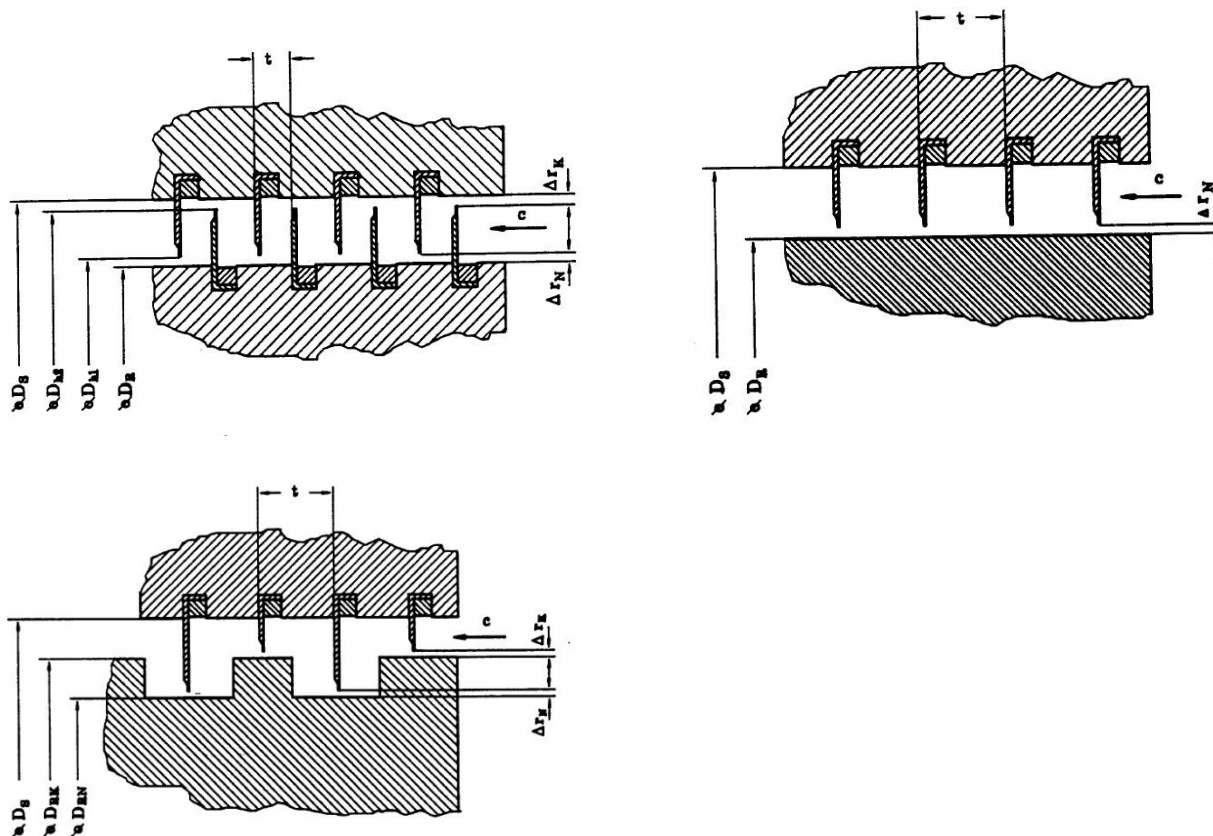
Cut through leading edge



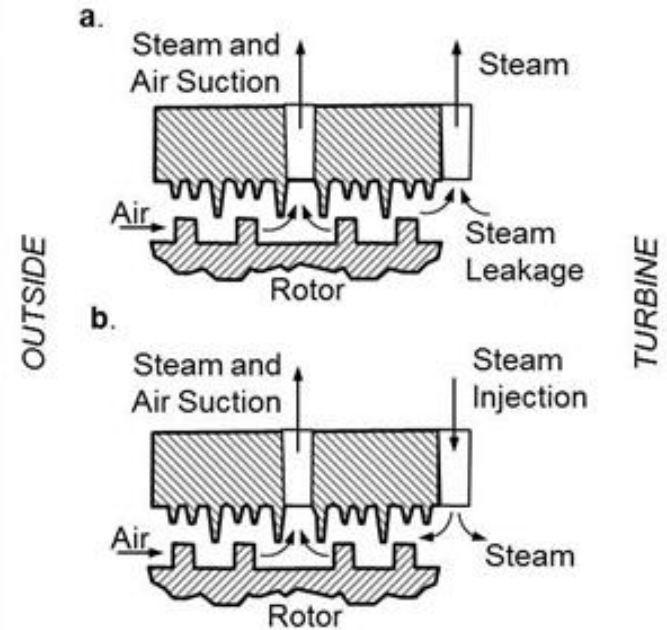
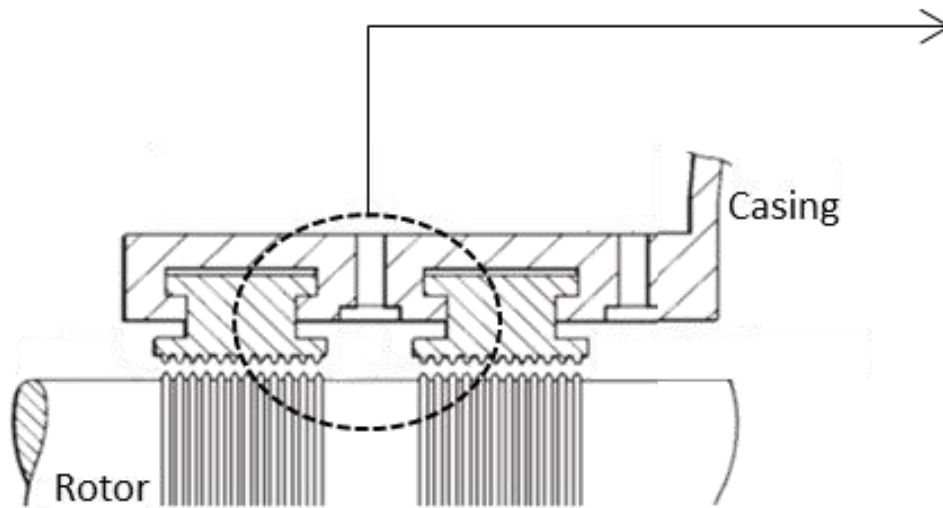
3. FINAL STAGES – low load limits



4. SEALS Seal designs

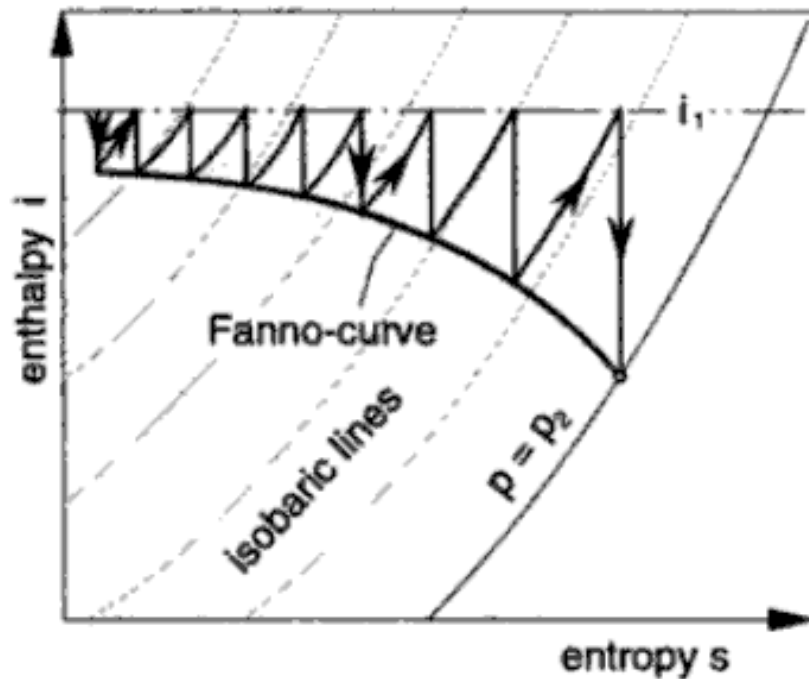


4. SEALS Operation

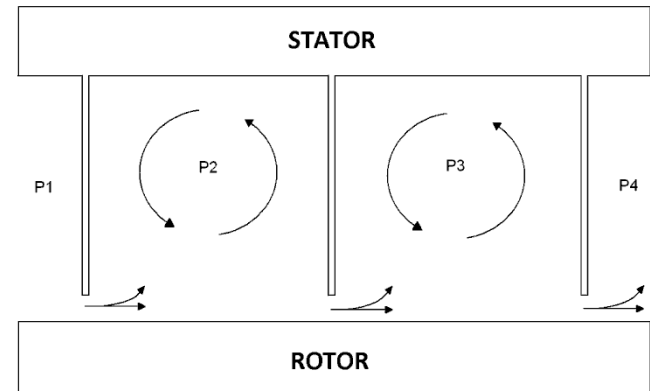


4. SEALS Theory

Fanno curve



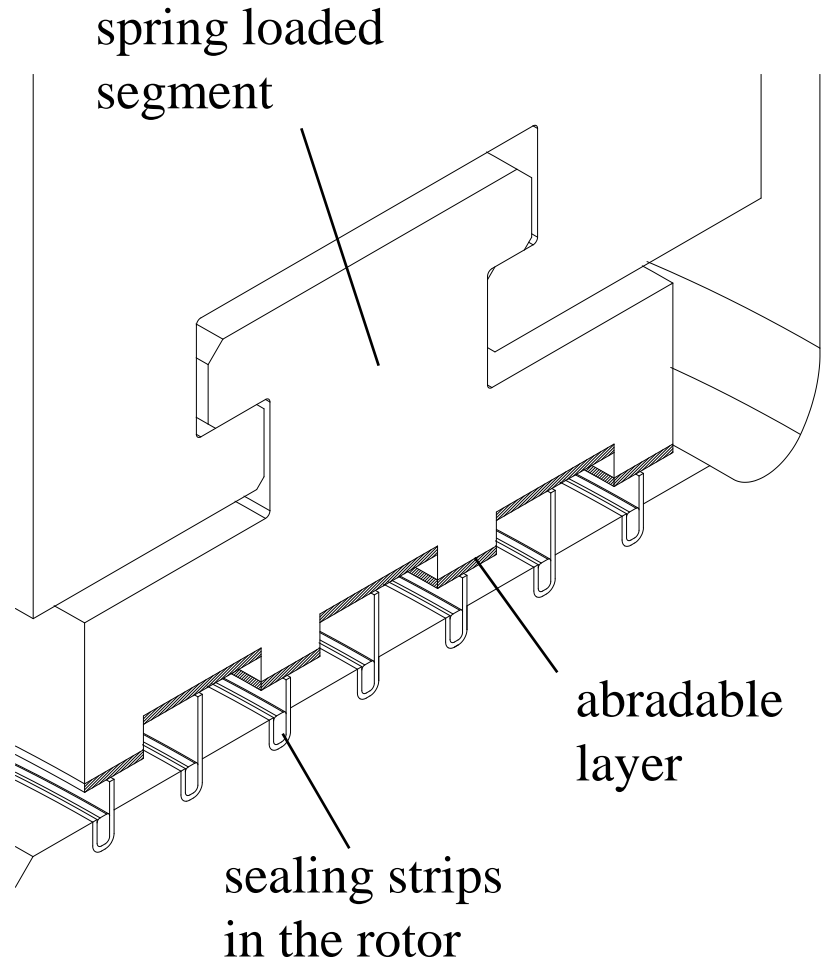
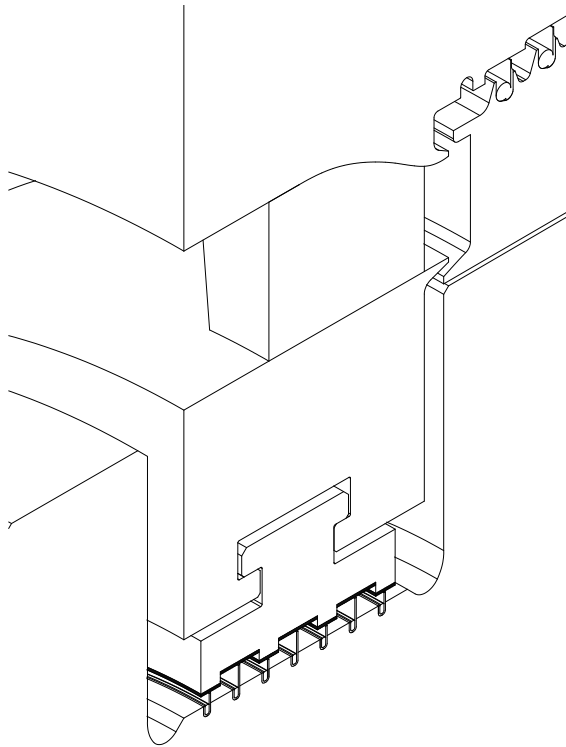
At clearance \rightarrow high speed



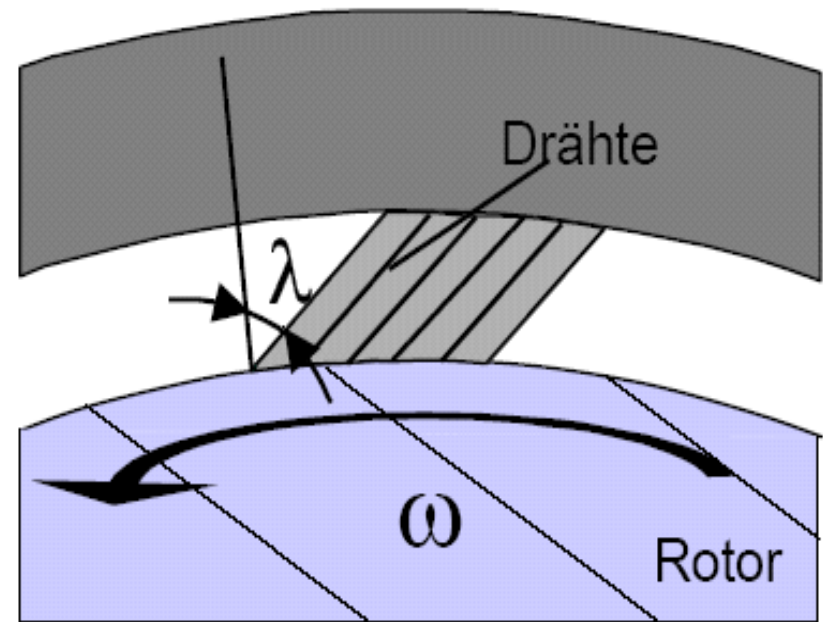
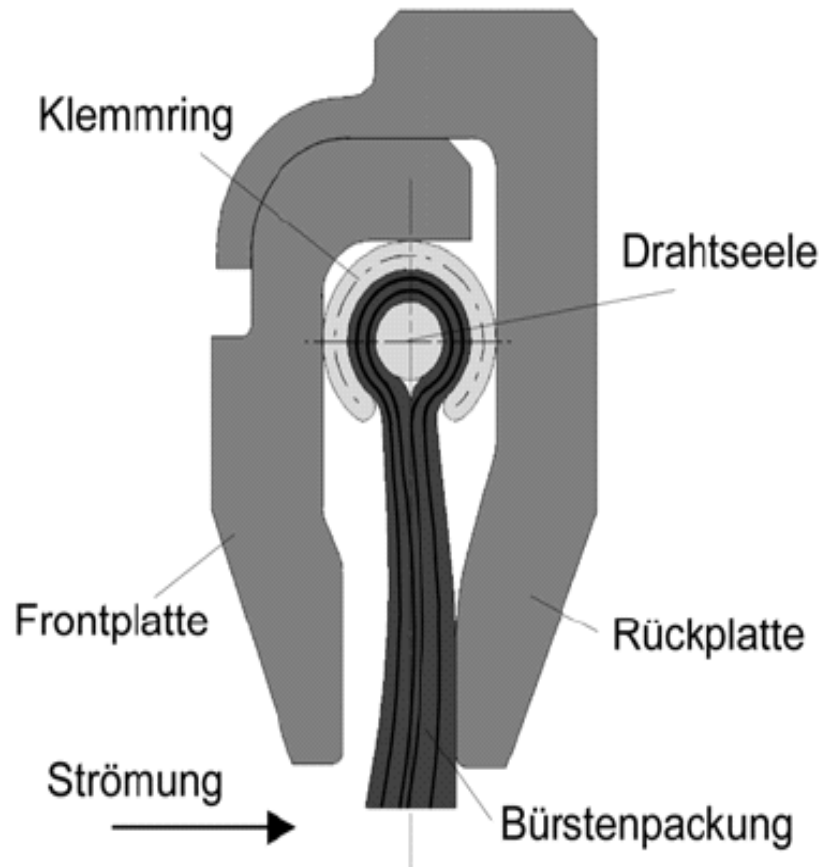
At cavity \rightarrow dissipated +
pressure decrease

4. SEALS

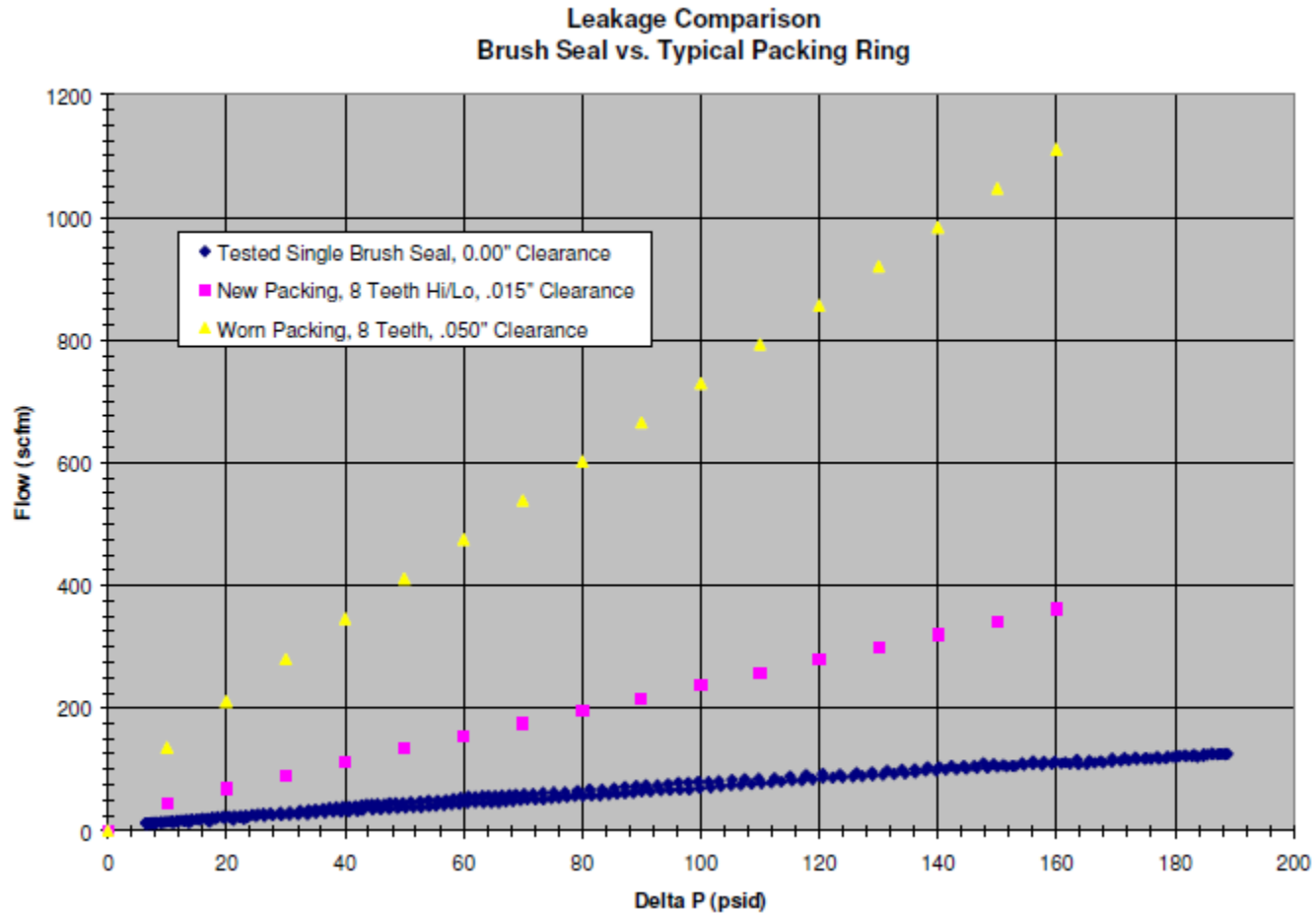
Abradable seals



4. SEALS Brush seal



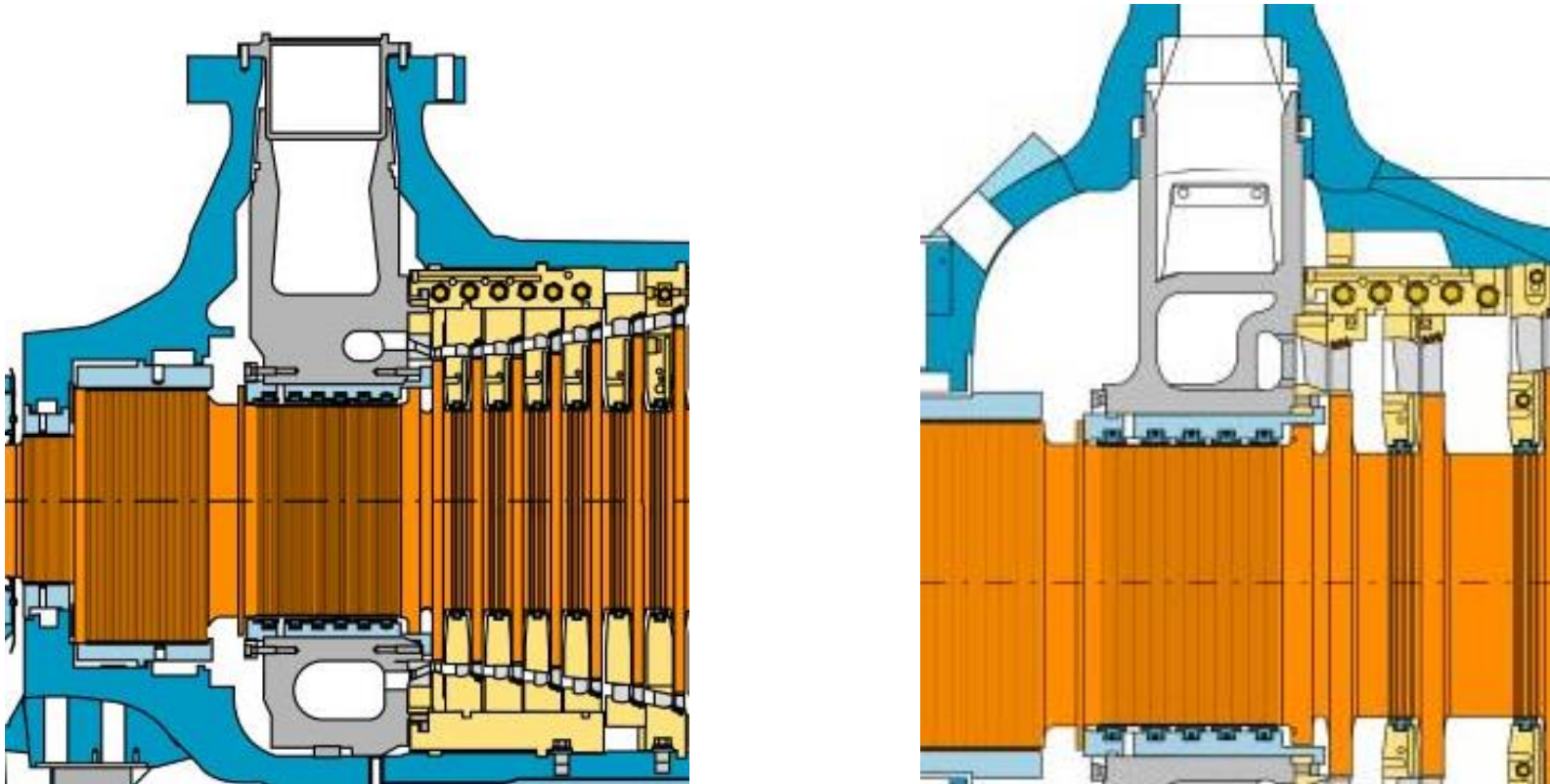
Seals comparison



[Turbocare, Power Gen 2005]

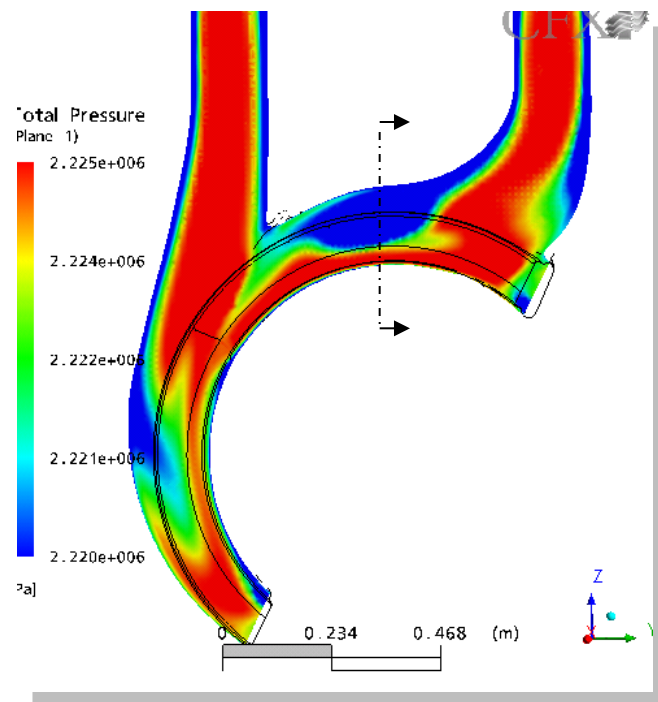
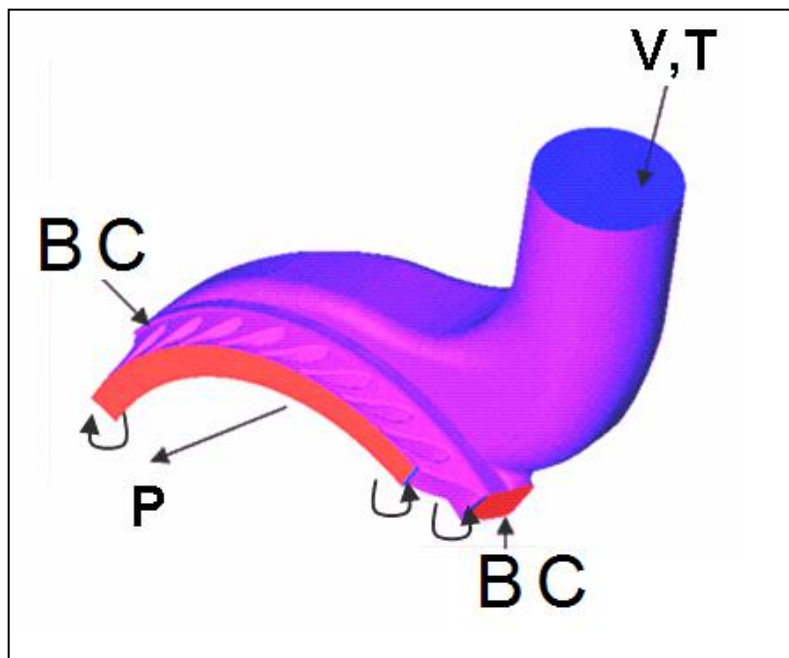
5. INLETS

Main function: admit steam



Inlet pressure losses

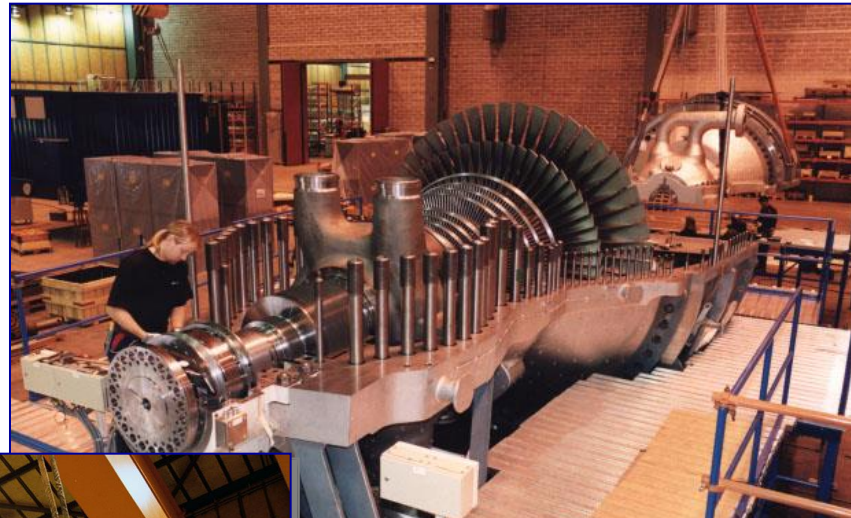
Figure of volute and flow



Volute Assembly



One 114 MW reheat unit in combined cycle.
Inlet 148 bara/ 2150 psia, 565°C/1050°F,
with reheat to 565°C/1050°F.
Order spring 1999.



One 95 MW reheat unit in combined cycle
with a GE 7FA gas turbine.
Inlet 129 bar(a)/ 1871psia and 568°C/ 1054°F
Order March 2000.



Three 112.7 MW units in combined cycle
with W501F gas turbines. Inlet data
per unit 112.7 MW at 145 bar(a)/ 2103 psia and
561.3°C/1004 °F.
Order 1998.

6. CASINGS

Main function: contain steam

Tightness

Mass

Thermal flexibility

6. CASINGS

Main function: contain steam

Important parameters

- Tightness
- Mass
- Thermal flexibility



6. CASINGS



Large pressure loads
High temperatures (creep
and LCF)

Welded designs and casted
designs

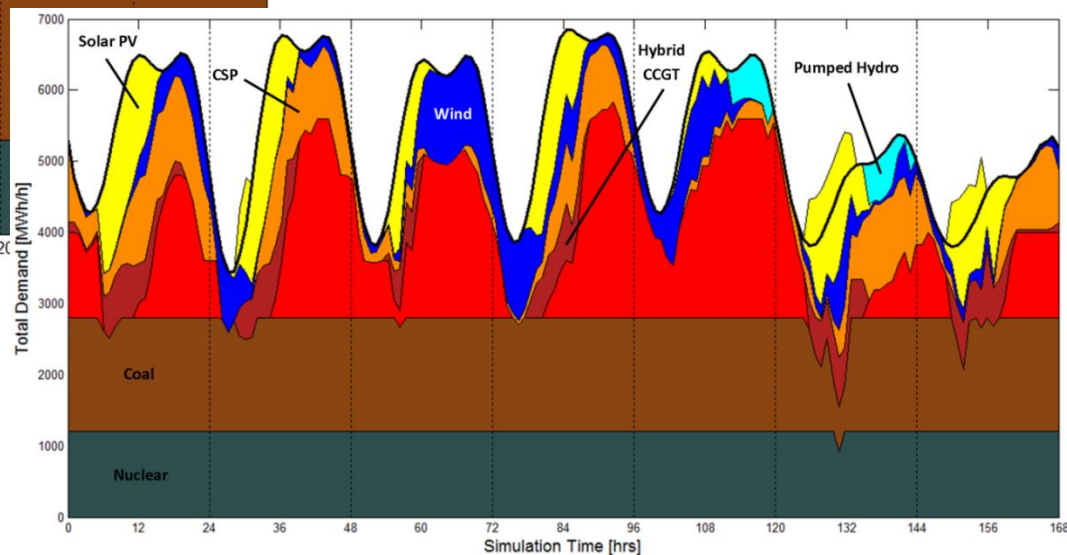
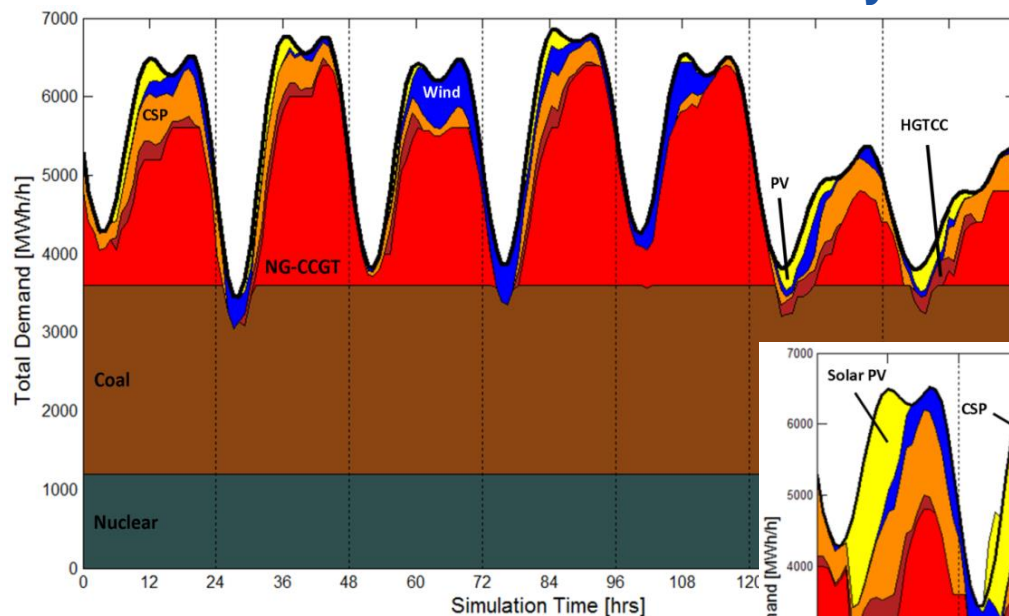
Modular design to adapt to
applications necessary

7. FLEXIBILITY

*Steam turbines have been available since the 19th century and nowadays are the **dominant** technology in electricity production.*



Electricity Market Changes



The Start-up Process

1) Initial Conditions

- Overnight standstill
- Pressure kept in drum overnight
- Turbine in turning gear and sealed

2) Boiler Start-up

- Recirculation to build pressure (mass flow)
- Steam conditions rising
- Allowed ramp rates $\left[\frac{K}{\min}\right] = f(P)$

3) Preheating

- Main header to turbine
- Bypass to condenser

4) Valve Opening

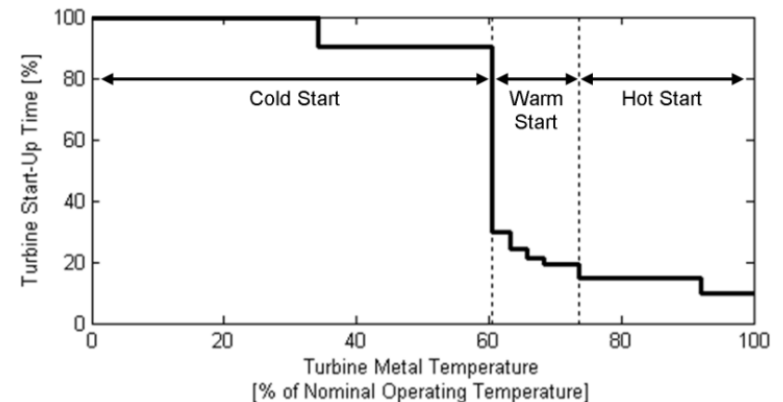
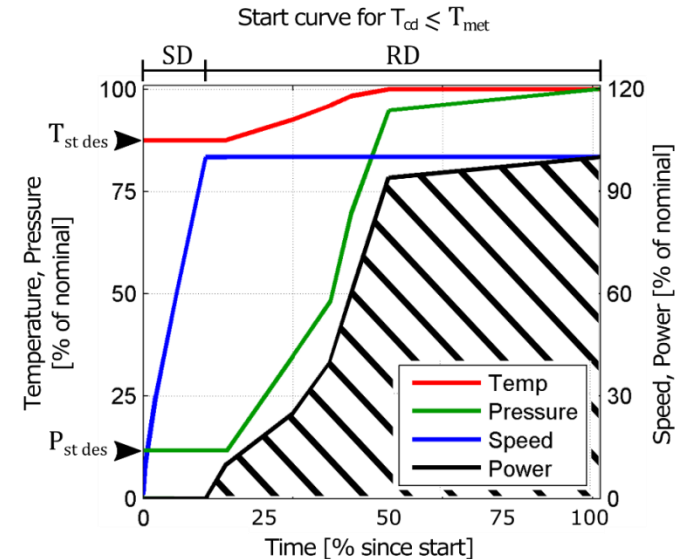
- Steam matching turbine requirements
- Pressure, Temperature, Desired superheat

5) Steam Turbine Start-up

- Temperature rates controlled

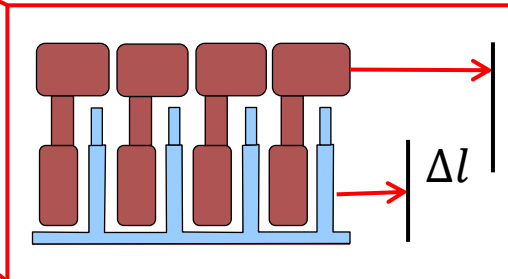
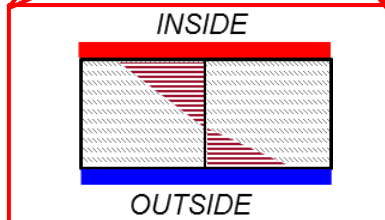
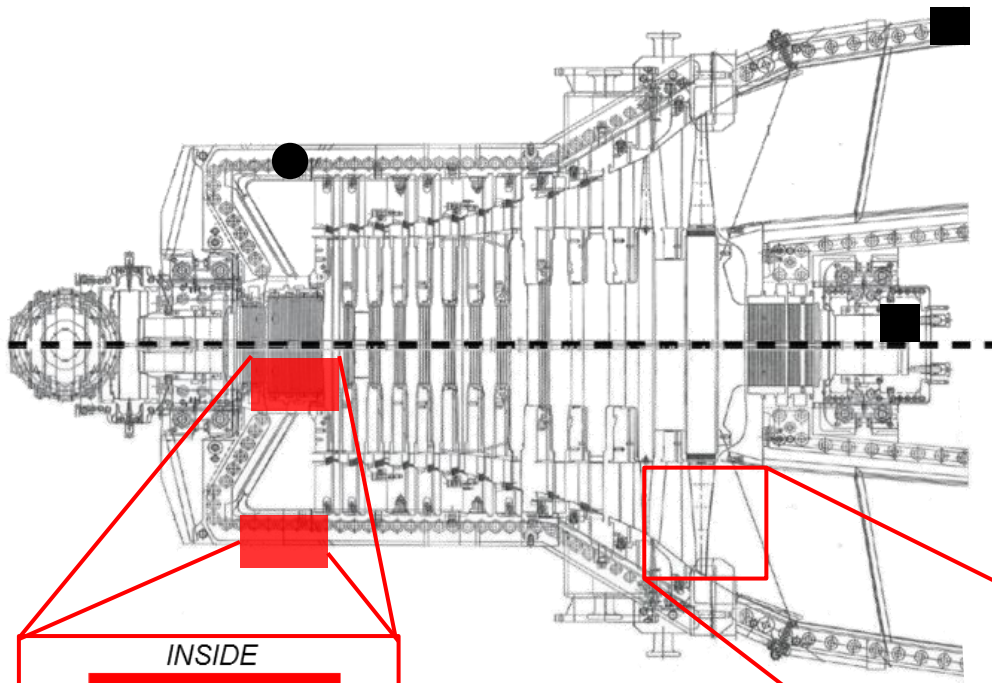
6) Sync+Load

- Rolling
- Loading



Steam Turbine Start-up Operation

- Steam temperature(t)
- $T_{\text{steam}} - T_{\text{metal}}$

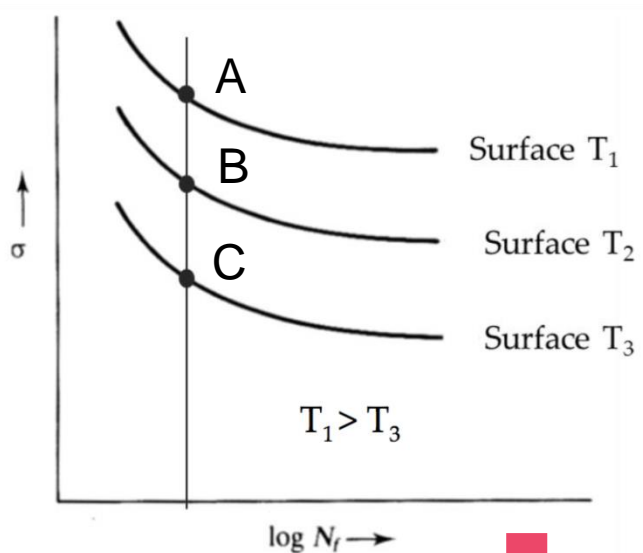


- **Thermal stress ●**
 - Thick-walled components
 - LCF Life
 - Allowed $\Delta T \rightarrow$ to last 40 yrs
 - Schedules/curves
- **Differential expansion ■**
 - Clearances (Leakage)
 - Rubbing
 - Monitoring

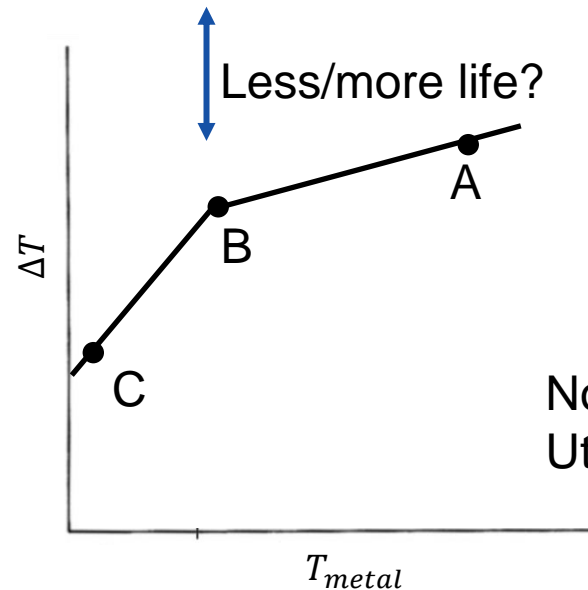
Thermal Stress and Life - Materials

$$\sigma = \frac{E\beta_m\Delta T}{1-\nu}$$

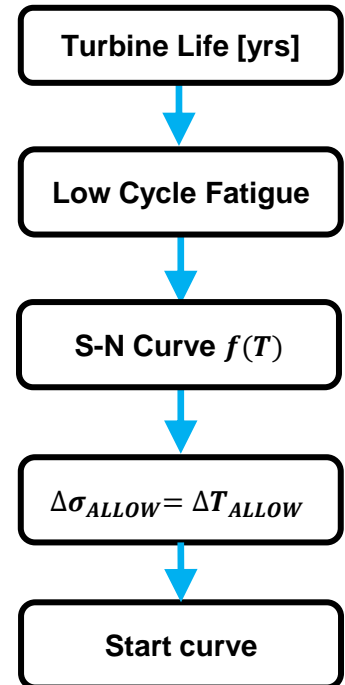
$$\Delta T = T_s - \overline{T_m} = T_s - \frac{2}{R^2} \int_0^R rT(r)dr$$



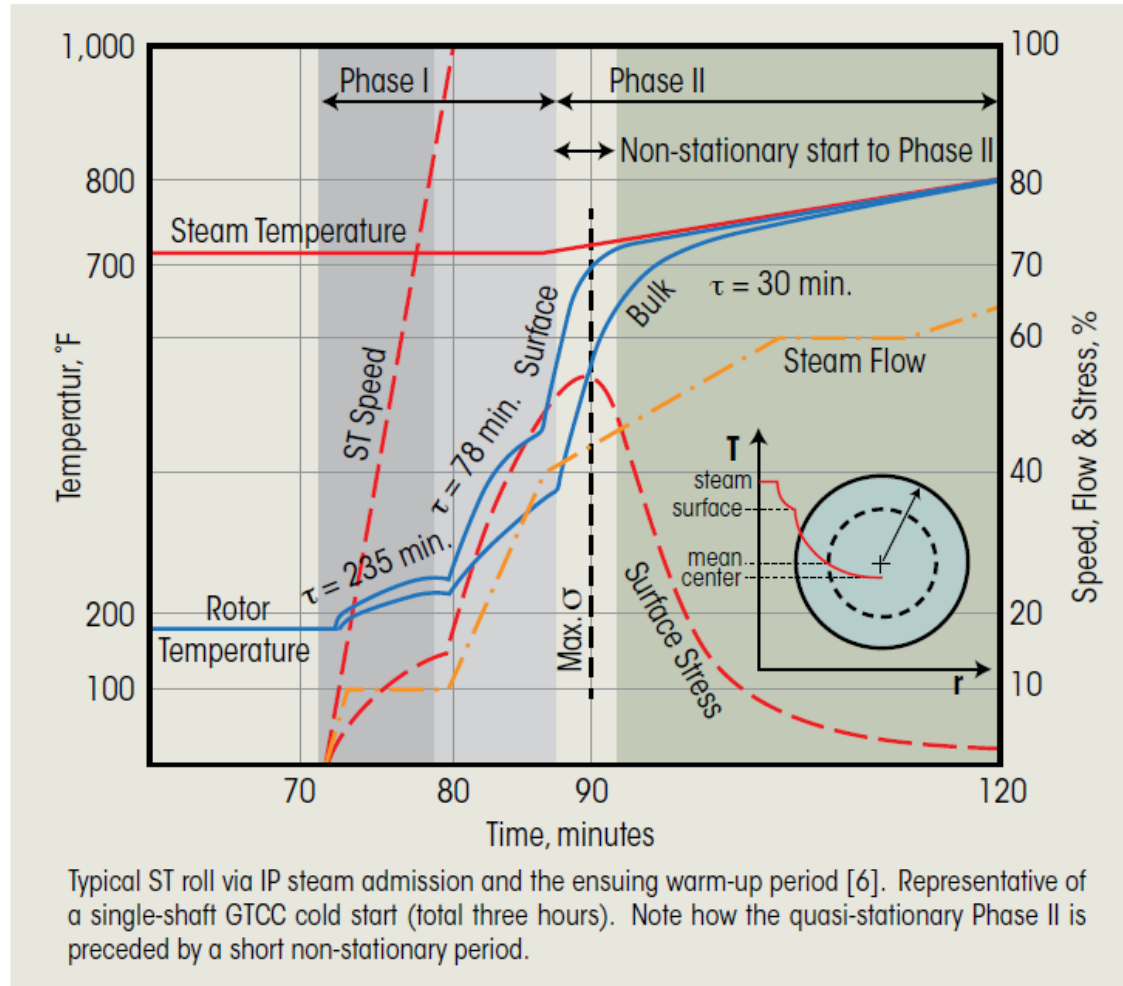
HS more life!



Notch factors!
Utilization Max



Thermal Stress and Life - Transient

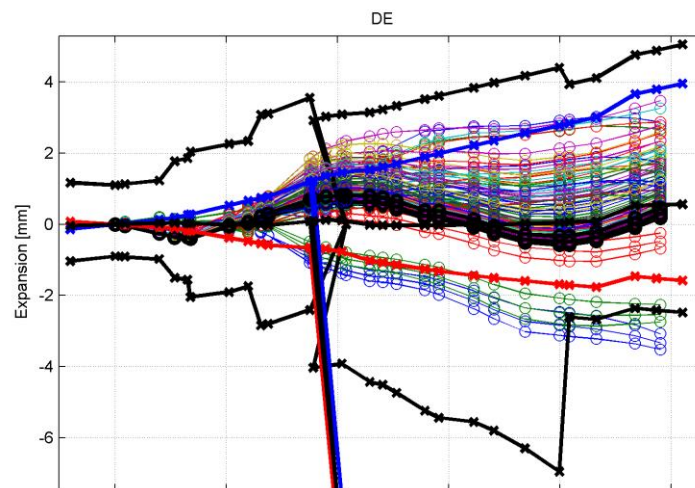
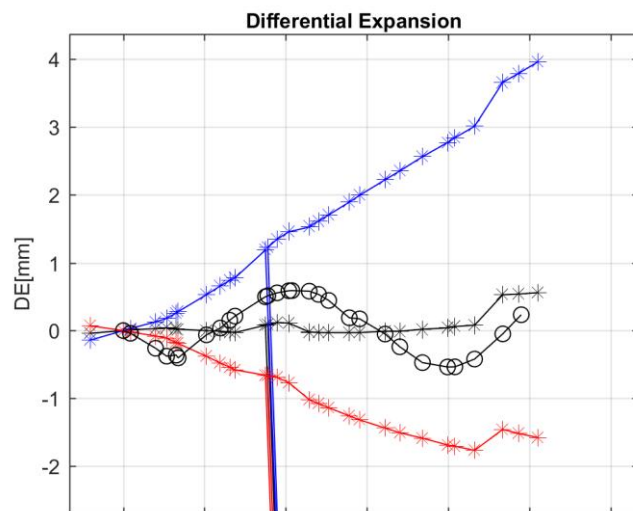
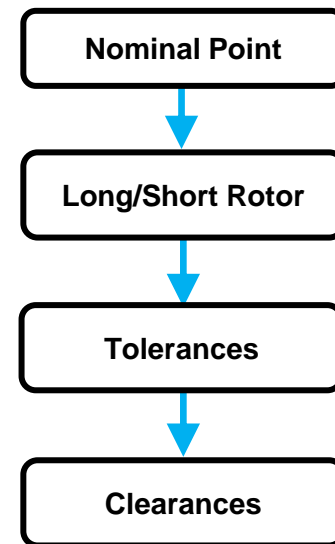
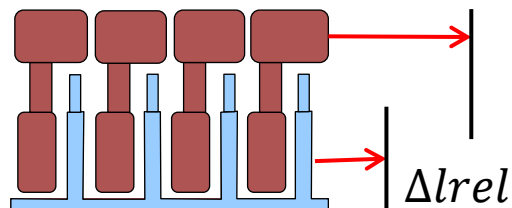


*Gülen "Gas Turbine CC Fast Start: The physics behind the concept"

Differential Expansion and Clearances

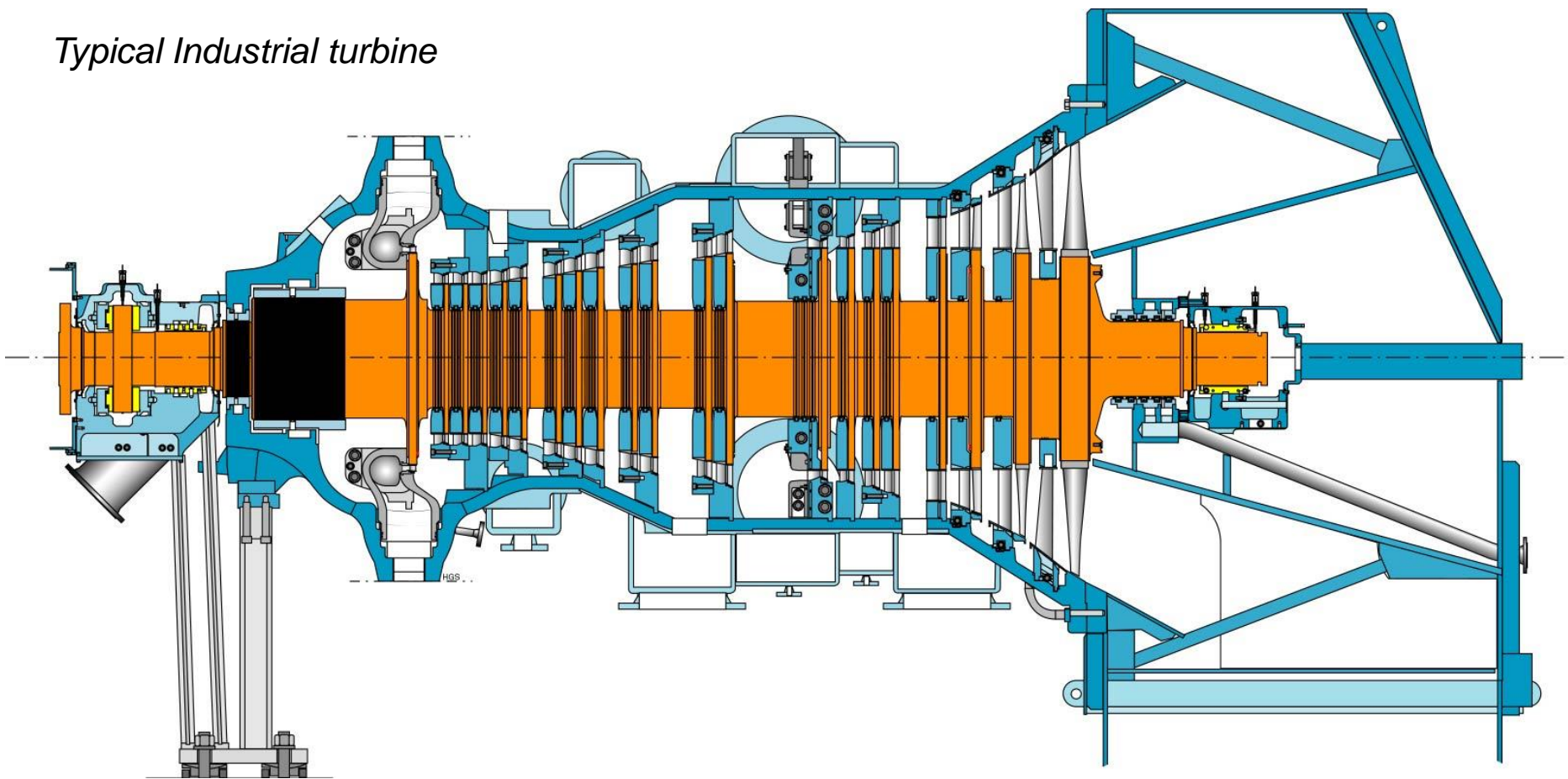
$$\Delta l = \beta_m l_i \cdot (T - T_i)$$

$$\Delta l_{rel} = \Delta l_{rot} - \Delta l_{cas}$$



Typical Turbine Layouts

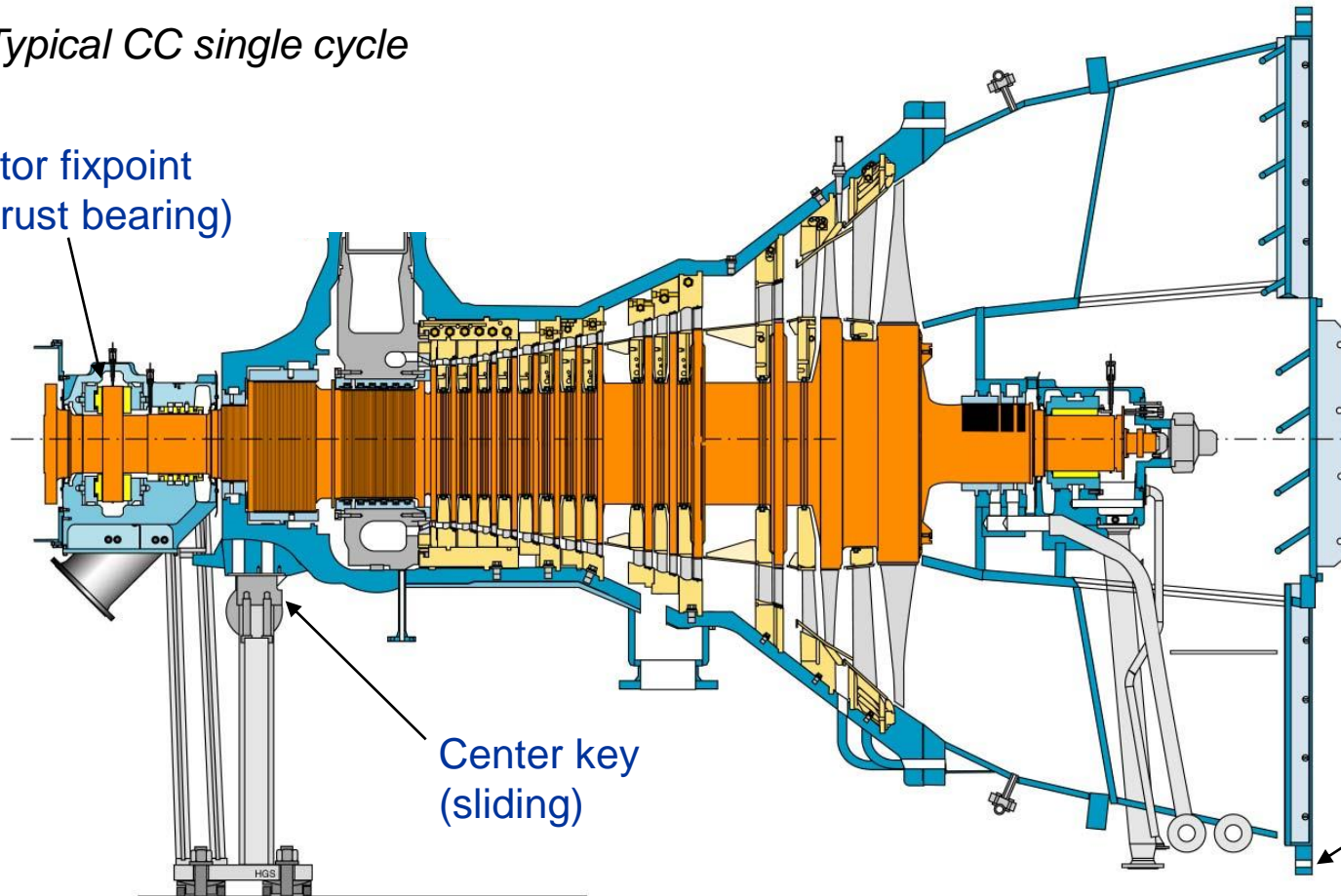
Typical Industrial turbine



SST900-SV-2-C500

Typical CC single cycle

rotor fixpoint
(thrust bearing)



Center key
(sliding)

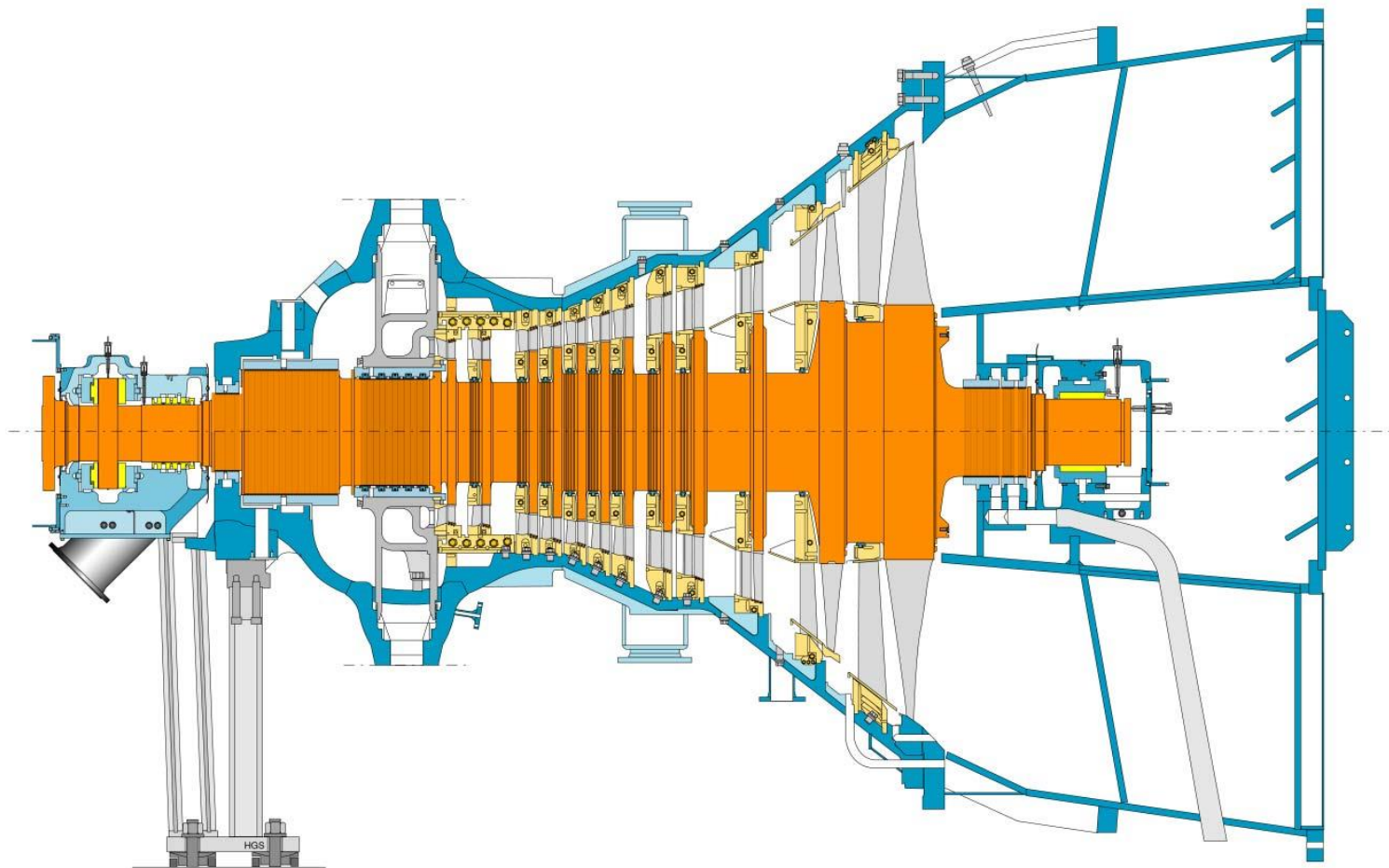
Flexible supports (no
sliding elements)

- Minimised clearances in HP-part
- No major sliding elements (maintenance)
- Cost effective
- High access for maintenance
- Compact
- Cost effective

casing fixpoint
(600mm in to
condenser)

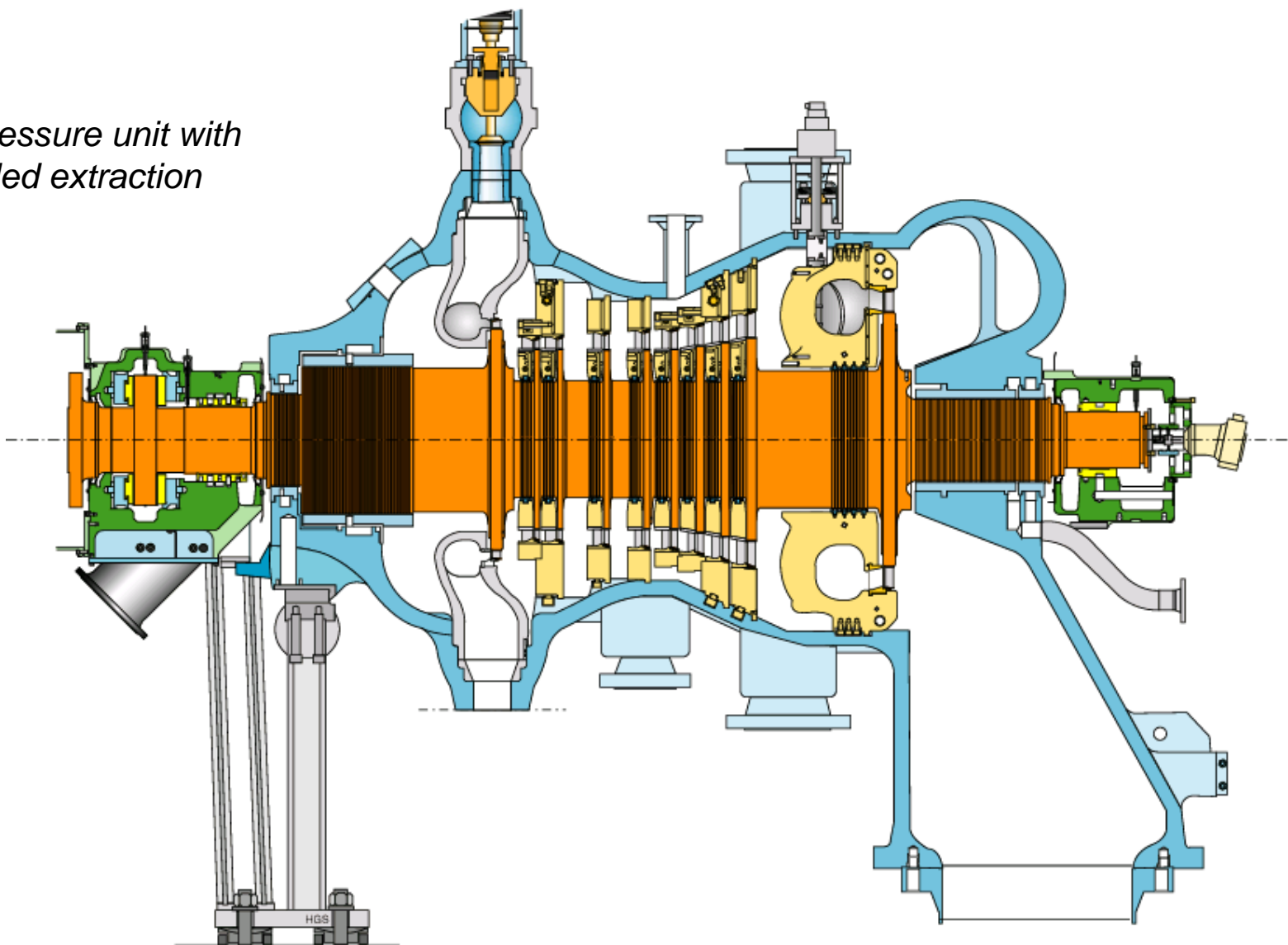
SST900-FV-2-C620

Typical IP for CC Reheat

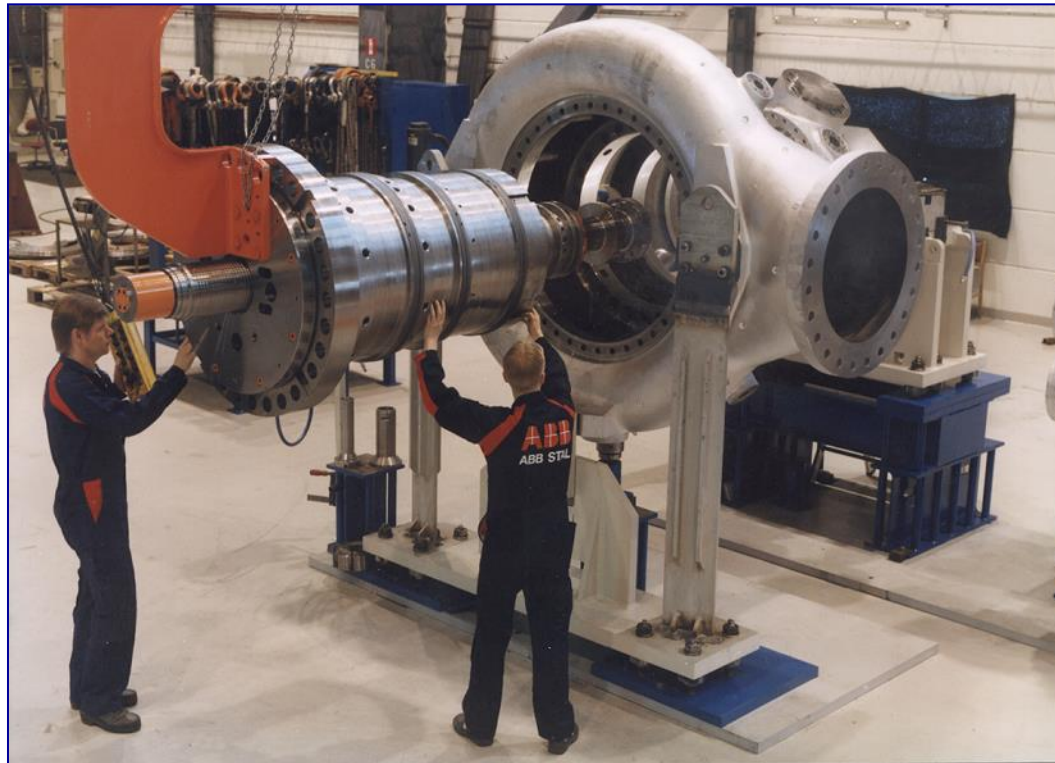


SST900-FN-25B-BE(down)

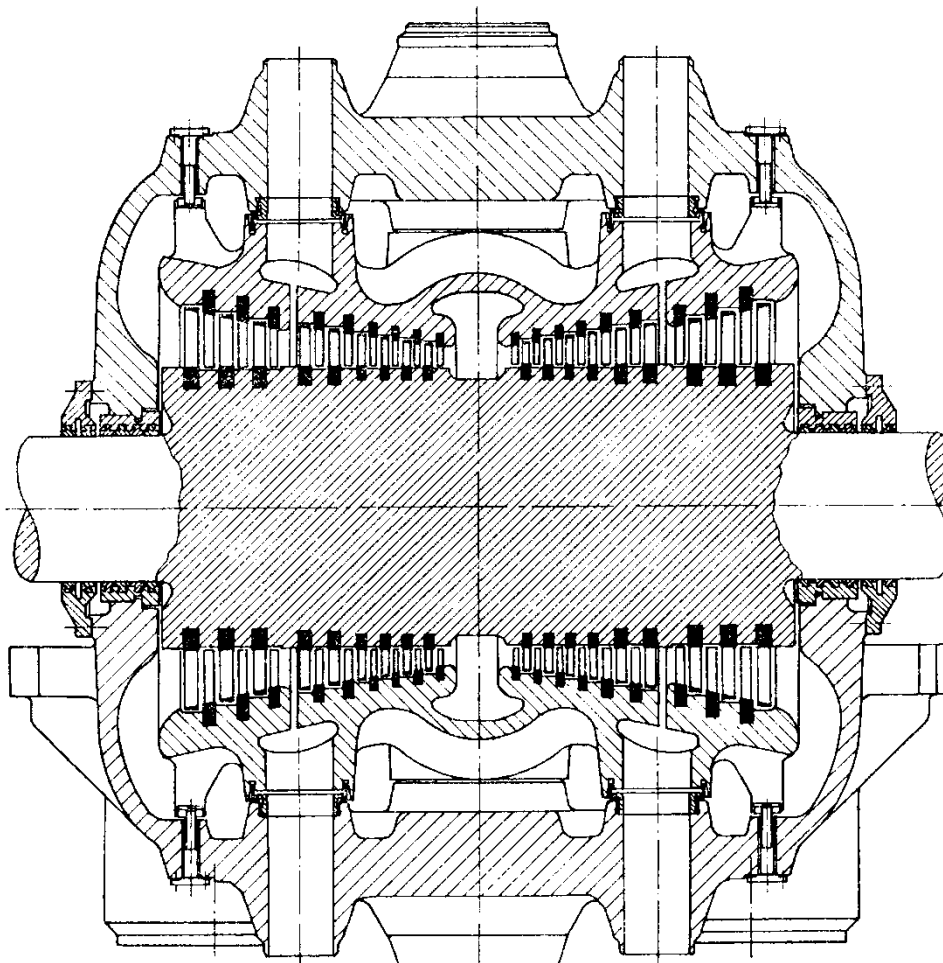
*Backpressure unit with
controlled extraction*



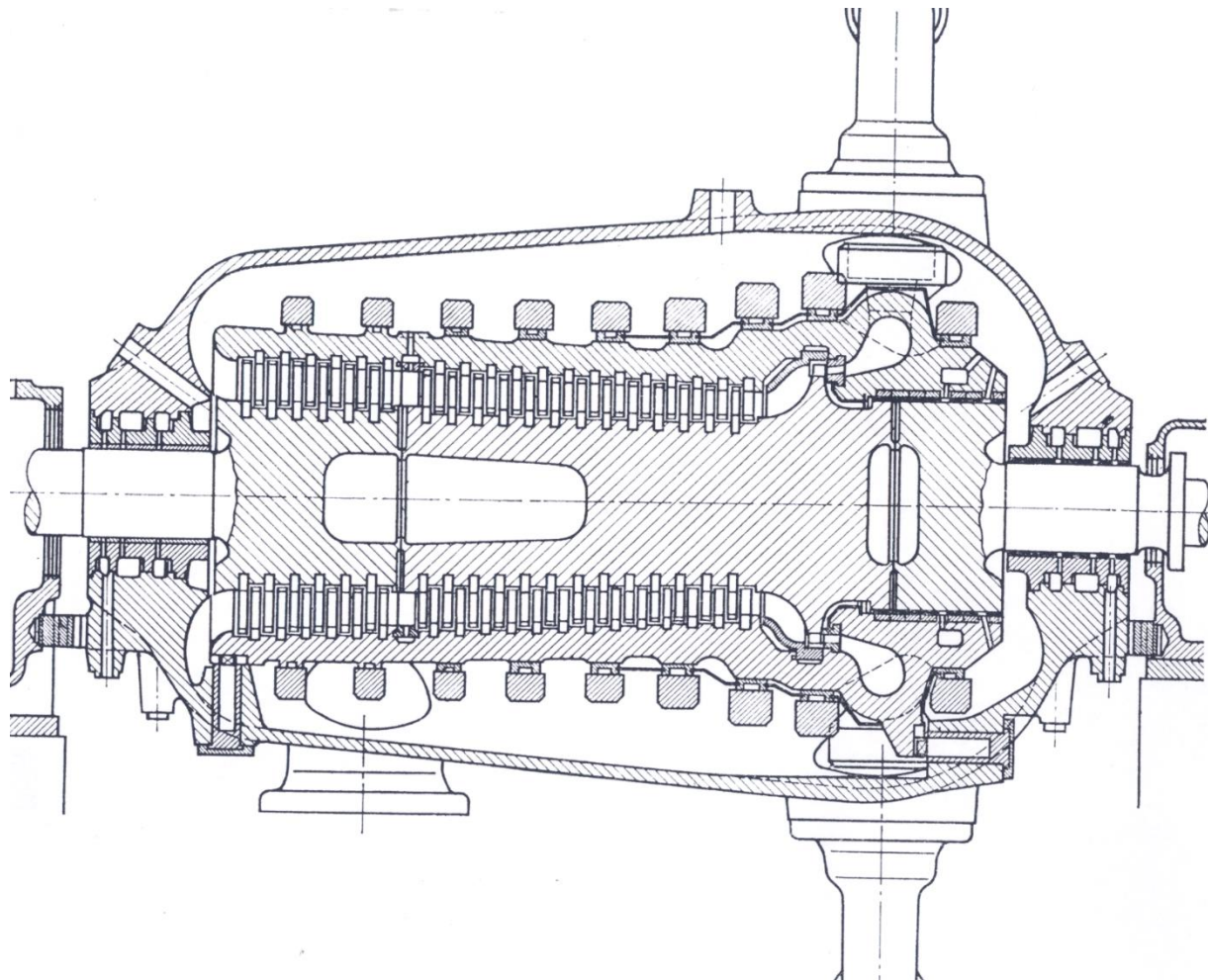
Siemens SST 700 HP-turbine – barrel design



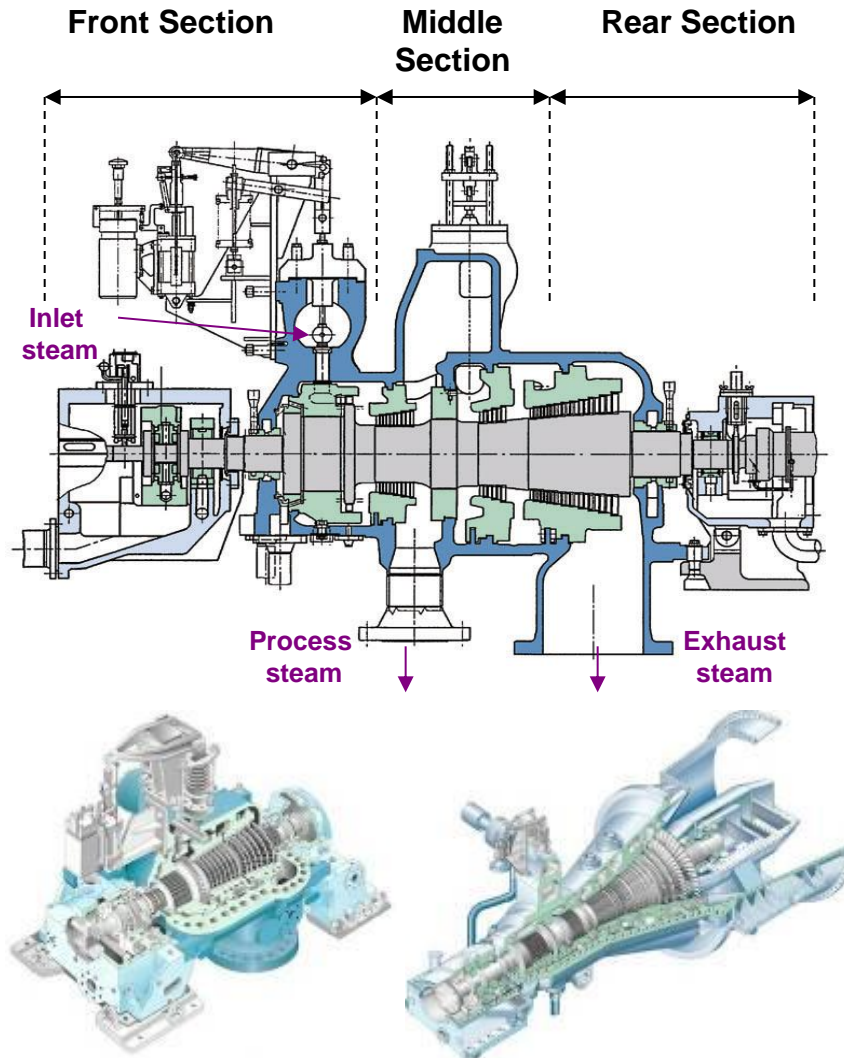
Dual flow high pressure steam turbine



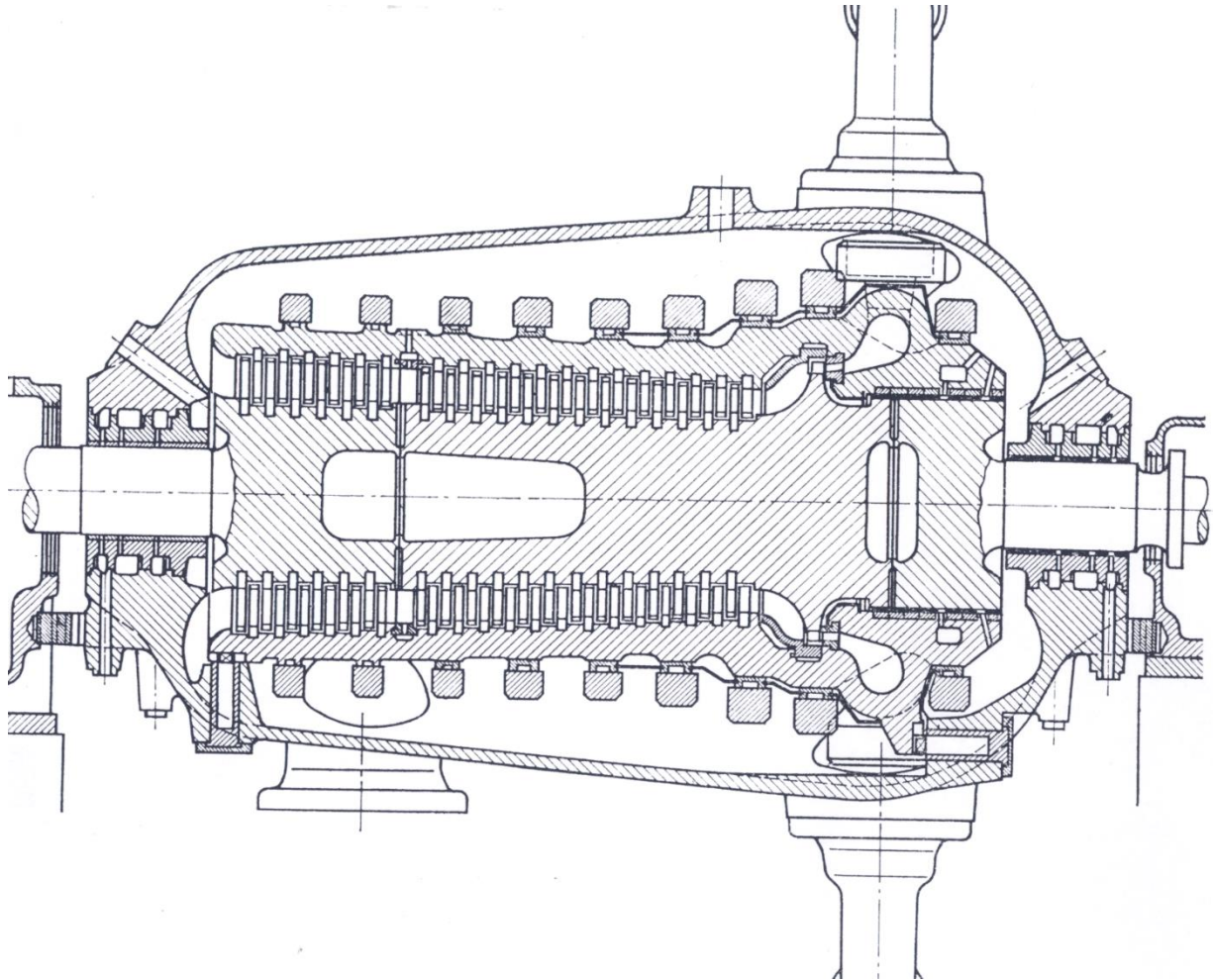
High pressure steam turbine, drum rotor



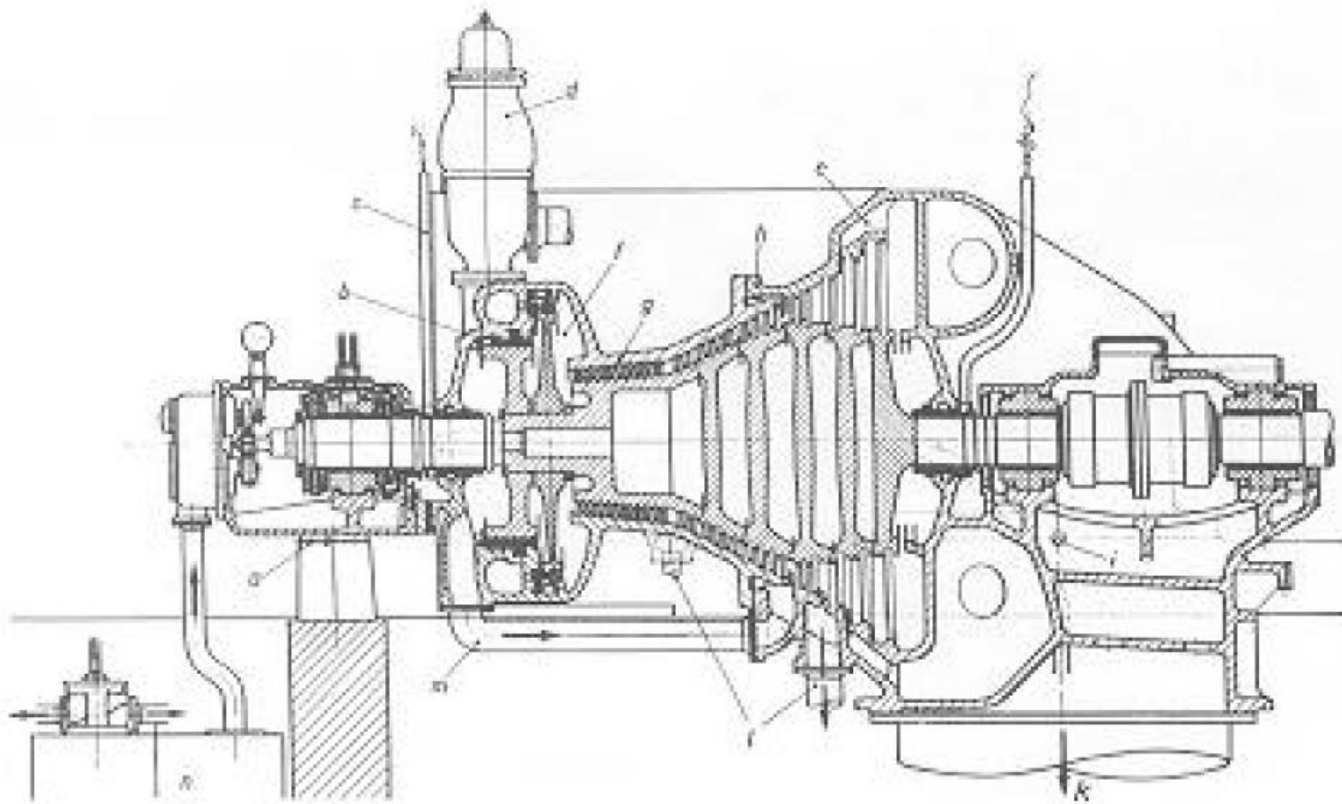
Steam turbine with controlled extractions



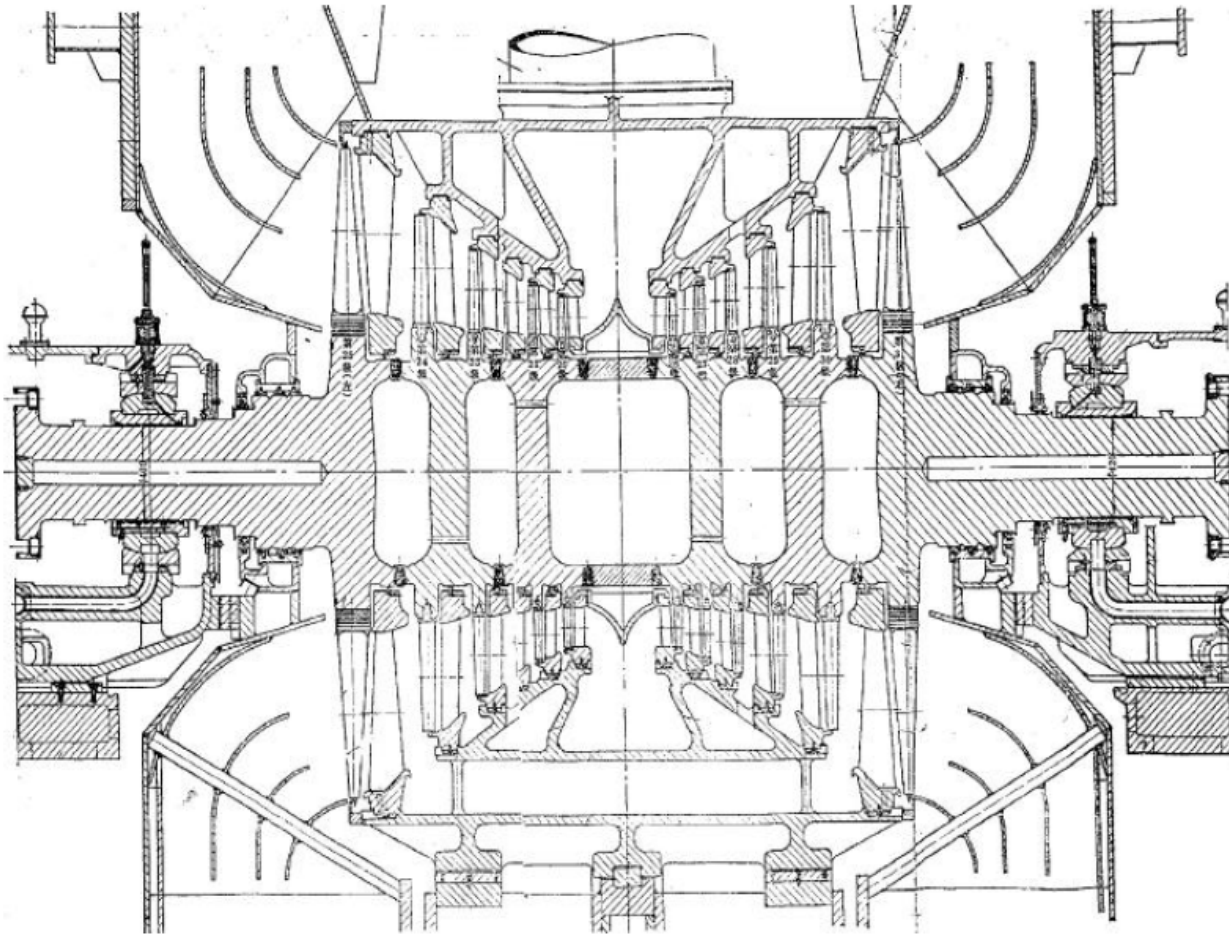
High pressure reaction steam turbine with control stage



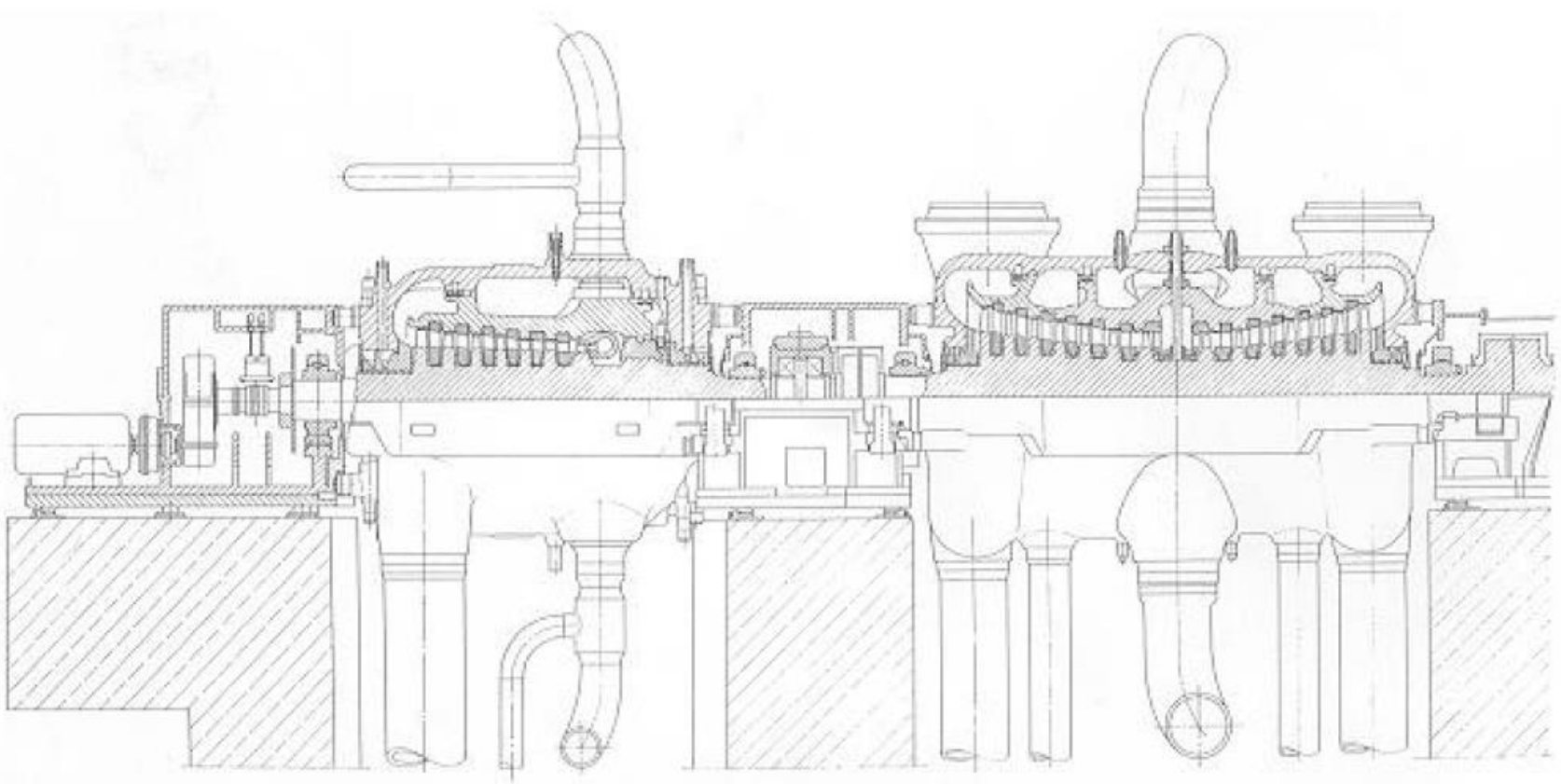
Reaction steam turbine



Low pressure impulse steam turbine from LMZ

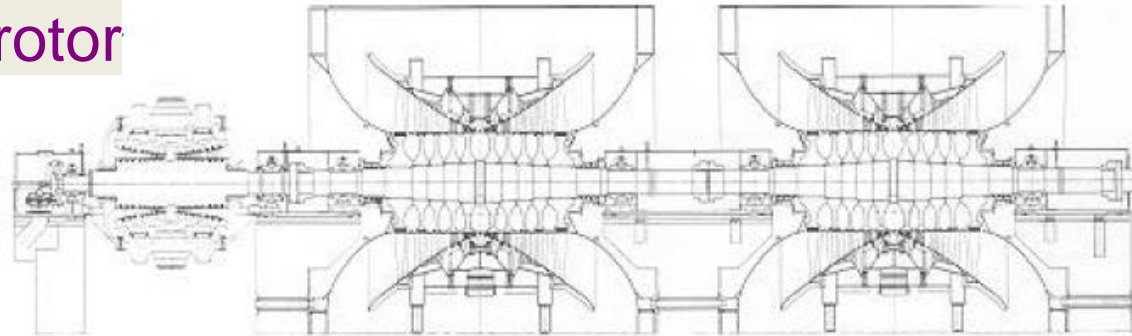


High and intermediate pressure steam turbine with impulse blading

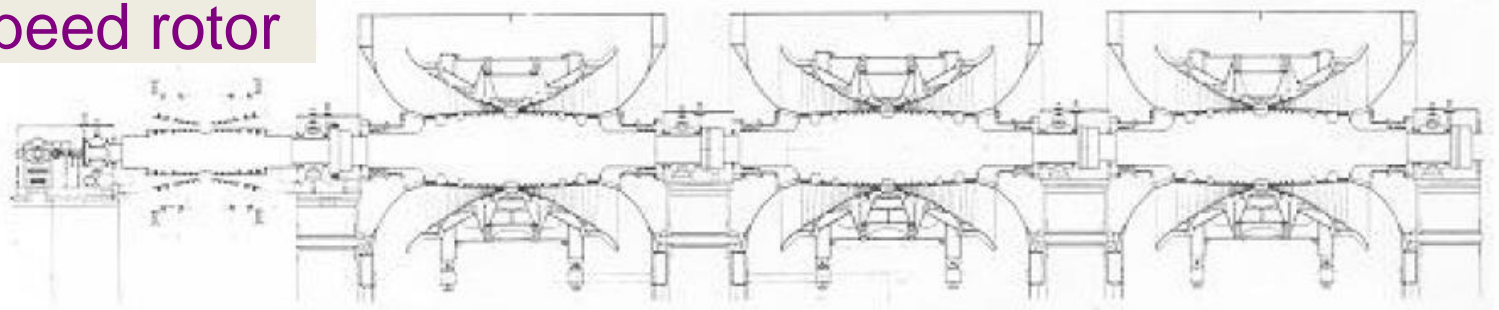


Typical nuclear power steam turbine trains

Half speed rotor



Full speed rotor





End of Presentation

Steam Turbine Technology

THANK YOU!